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ASSESSMENT OF FLEXIBLE LINES

FOR

FLOW INDUCED VIBRATION



UNLESS OTHERWISE SPECIFIED DIMENSIONS ARE IN INCHES TOLERANCES ON: FRACTIONS DECIMALS ANGLES		ASSESSMENT OF FLEXIBLE LINES FOR		SPACE FLIGHT CENTER NATIONAL AERONAUTICS
MATERIAL 10, 7 mbl. Significance 5/9/13 HEAT TREATMENT	SUBMITTED	FLOW INDUCED VIBRAT	TON	AND SPACE ADMINISTRATION HUNTSVILLE, ALABAMA
FINAL PROTECTIVE FINISH	APPROVED POLLE S/11/73	SCALE UNIT: WT	DWG SIZE	20M02540

REVISION LOG

	Revision Letter	Date of Revision	Revised Pages	Description
	Baseline	4/17/73		
Ī	A	5/14/73		
	В	9/25/79		g sale versuer in the second second
	С	8/18/87	ALL	Completely revised to reflect new analytical procedures developed in NASA TM-82556.
0 4	D D	2/28/90	ALL	Completely revised. Major changes include convolute bending mode, new examples, new computer program (ver. 3.2), new FNCO eqn., added safety factors, added oper. velocity criteria, corrected static stress eqns., added modified Goodman method, and general clarification.
72	E	12/19/91	3-11,14, 15,26,29, 33,34,36, 37,39,44, 40,41,45, 46,48-58, 61,63-69, 72,74,75, 77-86,90, 93-100, 103-106, 108, 110-114, 116-118	Modified scope of the document. Changed "safety factor" to "uncertainty factor." Deleted static stress eqns. (App. B). Added a system level analysis (para. 3.0). Deleted materials data (Tables 2 & 3). Changed method of fatigue assessment. Deleted modified Goodman approach. Added two references. Made minor changes to computer program. Made several other minor changes. NOTE: Refer to EO #2 for detailed description of changes.

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1.0 GENERAL

It is well known that the occurrence of flow-induced vibrations in flexible lines, specifically metal bellows and flexhoses, can cause premature failure. This is attributed to a resonance caused by the coupling of vortex shedding from the convolutes with the natural frequencies of the flexible line. A goal in designing these bellows and flexhoses is to prevent resonance from occurring. In the event this goal cannot be met, it is then desirable to analytically predict what the expected life of the bellows and flexhose is due to flow-induced vibration loads.

1.1 Scope

The purpose of this document is to establish the analytical methods for determining whether a given design of an annular convoluted metal bellows or flexhose is susceptible to flow-induced vibrations. These analytical methods include predicting the excitation flow range, frequency, and the corresponding stress resulting from only flow-induced vibration loads. This then leads to prediction of the expected life of the bellows or flexhose, with a final objective of achieving a theoretically infinite life for flow-induced vibrations.

The analytical assessment in this document shall be performed on all flexible lines consisting of formed annular

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convoluted metal bellows or flexhose, except those contained in paragraph 1.2, regardless of fluid velocity. It does not consider other bellows or flexhose configurations such as welded disc, ring reinforced, toroidal, etc. For those type configurations which do not fit this analysis, some other approved analysis or testing must be done.

The analytical model does not account for changes in the flexible line during thermal transients. Therefore, the assessment shall be performed three times on each flexible line in a application where its length changes as follows: First, for the flexible line in its free length; second, for the flexible line in maximum thermal compression; third, for the flexible line in maximum thermal extension.

The analytical method in this document was developed only for metal bellows and flexhoses manufactured with formed annular convolutes, as shown in Figure 2. These are the most commonly used type in propellant systems. The analytical model was developed in reference 1. The equations in reference 1 were empirically derived from extensive testing and are the basis of this document.

<u>CAUTIONARY NOTE:</u> The analysis in this document was developed for normal flexible line installations. It does not allow for installations where unusual flow disturbances exist

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(except for elbows located upstream of the flexible line) or for multi-phase flows.

CAUTIONARY NOTE: This document is intended as a tool for analyzing only one portion of the total design of a flexible line. The engineer must consider all possible load sources other than flow-induced vibration when determining the total system life of the flexible line (see paragraph 3.0). The engineer must also consider other requirements (stability, pressure capability, etc.) not covered by this document in the design of a flexible line.

1.2 <u>Excluded Flexible Line Assemblies</u>

- A. Instrumentation flexible lines.
- B. Flexible lines with steady-state flow of less than one second duration.
- C. Flexible lines with liners and sliding joints.
- D. Components which do not fall into any of the following flight criticality categories:
 - I. Personnel hazard
 - II. Mission/vehicle loss
 - III. Launch delay

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1.3 <u>Media</u>

Design analysis shall be repeated to verify the flexible line design integrity for all media imposed on the line, such as when a substitute medium is to be used in ground system checkout or other flow tests.

1.4 <u>Design Criteria</u>

There are two design criteria in which the flexible line (bellows and flexhose) shall be designed to meet. These are listed below:

- 1. The flexible line shall be designed to meet a theoretically infinite life, if its high cycle material curve exhibits a true endurance limit, for flow-induced vibration loads at its expected operating conditions. For a material not exhibiting a true endurance limit, an endurance limit as defined by each program shall be used.
- 2. The maximum operating flow velocity of the flexible line shall be limited per paragraph 2.6.

2.0 DESIGN ANALYSIS PROCEDURE

The procedure for analyzing a given bellows or flexhose configuration for susceptibility to fatigue failure

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from only flow-induced vibration loads consists of several different steps as follows:

For a Bellows:

- Step 1. Calculate the natural frequencies for all vibration modes of the bellows: longitudinal modes and local convolute bending mode (see paragraph 2.1.1).
- Step 2. Calculate the flow excitation velocity range for each mode of vibration (see paragraph 2.2).
- Step 3. If the flow medium is a gas, calculate the first radial acoustic mode frequency and velocity (see paragraph 2.3).
- Step 4. Calculate the flow-induced stress for each mode of vibration defined in step 1 and apply the appropriate uncertainty factors (see paragraph 2.4).
- Step 5. Determine whether infinite life is achieved (see paragraph 2.5).
- Step 6. Determine the maximum operating velocity limit (see paragraph 2.6).

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For a Flexhose:

- Step 1. Calculate the natural frequencies for the three vibration modes of the flexhose:

 in-phase longitudinal mode, out-of-phase longitudinal mode, and local convolute bending mode (see paragraph 2.1.2).
- Step 2. Calculate the flow excitation velocity range for each mode of vibration (see paragraph 2.2).
- Step 3. If the flow medium is a gas, calculate the first radial acoustic mode frequency and velocity (see paragraph 2.3).
- Step 4. Calculate the flow-induced stress for each mode of vibration defined in step 1 and apply the appropriate uncertainty factors (see paragraph 2.4).
- Step 5. Determine whether infinite life is achieved (see paragraph 2.5).
- Step 6. Determine the maximum operating velocity limit (see paragraph 2.6).

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2.1 <u>Natural Frequency Calculation</u>

2.1.1 Free Bellows

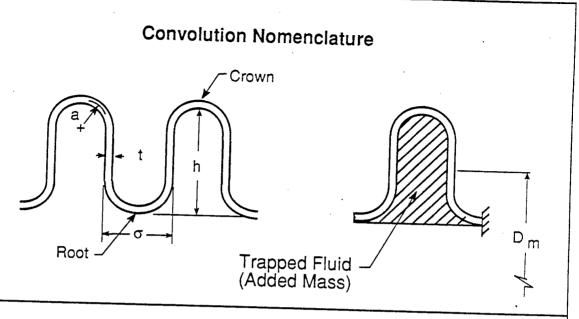
Consider the bellows structure represented by a lumped spring-mass mechanical model as shown in Figure 1. The pertinent bellows nomenclature used in the frequency calculation is given in Figure 2. All of the dimensions used in these calculations should be obtained by measuring the actual bellows being used. The user is advised that the asbuilt dimensions of a bellows can vary significantly from the specified drawing dimensions. As determined in reference 1, this can cause significant differences in the final results.

Step A. Calculate the elemental spring rate of one-half of a convolution, k, from the expression:

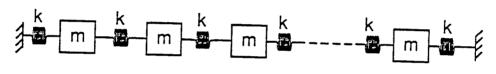
$$k=2N_{C}K_{a}$$
 (1)

where K_a is the overall bellows spring rate $\frac{determined}{determined}$ experimentally from a force-deflection test. For a new bellows assy., the user is required to employ experimental values obtained from a force-deflection test. For those bellows where a force-deflection test is not obtainable (i.e. a bellows already installed permanently in a line assy.), a





Mechanical Model and Nomenclature



A mass (m) is assigned each convolution crown and root; the number of masses is $2N_{\rm C}$ -1. The value of m is m_m +m_f where

$$m_{m} = \frac{\pi \rho_{m} D_{m} t N_{\rho} [\pi a + (h-2a)]}{g}$$

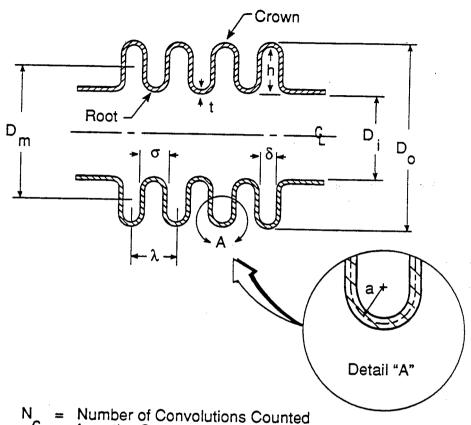
The number of springs is 2N_C and

where K_A is the overall bellows spring rate.

Figure 1. Lumped Spring-Mass Mechanical Model for Free Bellows

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from the Outside

Number of Plys

D_m = Mean Bellows Diameter

Wall Thickness (Thickness per Ply if Multi-Ply)

λ Inside Convolute Pitch

Inside Convolute Width σ

Mean Convolute Radius а

Mean Inside Convolute Height h

δ Inside Convolute Gap

Figure 2. Bellows Nomenclature

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rough estimate of $\mathbf{K}_{\mathbf{a}}$ may be made from the following expression:

$$K_a = D_m E (N_p/N_c) (t/h)^3$$
 (2)

where E is Young's modulus for the bellows material at the operating temperature. If the bellows is designed to operate in the plastic range of the material then an adjusted value of E should be used in the calculations throughout this spec. One suggested method for adjusting E is discussed in paragraph 3.0.

Step B. Calculate the elemental metal mass, $\mathbf{m}_{\mathbf{m}},$ from the equation:

$$m_{m} = \frac{\mathcal{Y} \mathcal{P}_{m} D_{m} t N_{p} [\mathcal{Y}a + h-2a]}{q}$$
 (3)

where \mathcal{P}_{m} = weight density of bellows material (lbf/in³)

g = gravitational acceleration

 $a = mean convolute radius = (\sigma - tN_p)/2$

h = mean inside convolute height

 $D_{\rm m}$ = mean diameter of bellows

t = ply thickness

 N_p = number of plys

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Step C. Calculate the elemental fluid added mass, $m_{\rm f}$, consisting of two types of loading, $m_{\rm fl}$ and $m_{\rm f2}$, given as follows:

$$m_{f1} = \frac{\mathcal{T} \mathcal{P}_f D_m h (2a - tN_p)}{2g}$$
 (4)

$$m_{f2} = \frac{p_f D_m h^3}{g \delta}$$
 (5)

where \mathcal{P}_f = weight density of fluid (lbf/in³) $\delta = \text{inside convolute gap} = \lambda - \sigma$

Now, the total elemental fluid added mass in slugs is given by the empirical equation:

$$m_f = K_1 m_{f1} + K_2 m_{f2} (N/N_c)$$
 (6)

where $K_1 = 1.0 \text{ (non-dim)}$

 $K_2 = 0.68 \text{ (non-dim)}$

 $N = \text{mode number} = 1,2,3,...2N_{C}-1$

Step D. Calculate the dimensionless frequency factor, $\mathbf{B}_{N}, \\$ for each mode number N from the equation:

$$B_{N} = \{2[1+\cos(180(2N_{C}-N)/2N_{C})]\}^{1/2}$$
 (7)

Alternately, the dimensionless frequency factor may be obtained from Table 1 for certain values of N.

Table 1. Dimensionless Frequency Factors $\mathbf{B}_{\mathbf{N}}$

MODE NUMBER N

L	1		2	3	4	5	6	7	8	9	10	11	12	13	14	15	16	17	18	19	20	21	22	23	24	25
1	1.41	4											 			\vdash	 		 '''	 	 	 -				
2	0.76	5 1.4	114	1.845			l							 		-	 	 	 	 	 	 	 	\vdash	 	├─
3	0.52	0 1.0	100	1.414	1.732	1.930			<u> </u>				l	 		 	 	 		 	 	 	 	 	 	├
4	0.39	0 0.7	65	1.111	1.414	1.663	1.848	1.962		 	 	 		 		 	 		 	 				 		
5	0.31	4 0.6	18	0.908	1.176	1.414	1.618	1.782	1.902	1.975	<u> </u>	 			 	 	 	 	 -		 		├			
6	0.26	4 0.5	18	0.765	1.000	1.217	1.414	1.587	1.732	1.848	1.932	1.983		 		 -	_			 	 	 		 		
7	0.22	6 0.4	45	0.661	0.868	1.064	1.247	1.414	1.563	1.693	1.802	1.888	1.950	1.987		<u> </u>	_			 	 					\vdash
8														1.913		1.990	 		 	 		 -	 	-		
9														1.812				1 992		<u> </u>		 	 			
10														1.705						1 993			 			
11														1.601								1 994				-
12														1.503										1 006		
13																									1.985	1 00

NOTE: The Dimensionless Frequency Factors were Determined from the Equation

$$B_{N} = \sqrt{2 \left\{ 1 + \cos \left[\frac{180(2N_{c} - N)}{2N_{c}} \right] \right\}}$$

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Step E. Calculate the reference frequency from the equation:

$$f_0 = \frac{1}{2\pi} \sqrt{\frac{k}{m}}$$
 (8)

where f_o = reference frequency

k = elemental spring rate

 $m = m_m + m_f = total elemental mass$

NOTE: For a free bellows there are two different kinds of structural modes which may be flow-excited. They are the longitudinal modes and the local convolute bending mode. These are illustrated in Figure 3.

Step F. Calculate the true longitudinal mode frequencies for each mode number N from the equation:

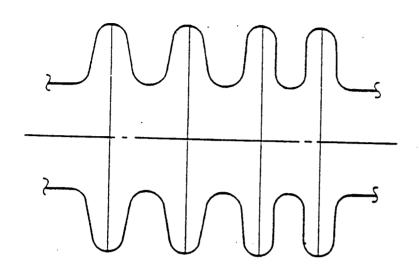
$$f(N) = (f_0) (B_N)$$
 (9)

where f(N) = modal frequency (Hz)

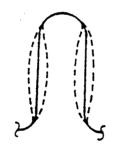
Step G. Calculate the local convolute bending mode frequency from

$$f_{CB} = \frac{1}{2\pi} \sqrt{\frac{8k}{m_{m} + .68m_{f2}}}$$
 (10)

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Axisymmetric Longitudinal Modes



Higher Order Local Convolute Bending Mode

Figure 3. Summary of Bellows Vibration Modes

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2.1.2 Flexhose

Consider the convoluted hose structure represented by the lumped spring-mass mechanical model as shown in Figure 4. Note that for a flexhose the value $N_C=1$ will be used.

Step A. Calculate the elemental spring rate of one-half of a convolution, k, from the expression:

$$k = 2K_{f} \tag{11}$$

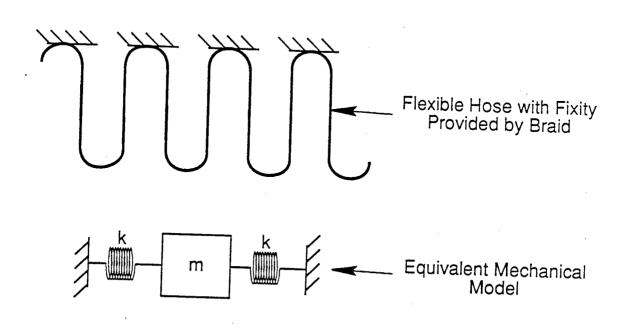
where K_f is the spring rate <u>for one complete</u> <u>convolution</u> (N_C =1) determined experimentally from a force-deflection test. Note that the overall spring rate obtained from test must be multiplied by the actual number of convolutes in the hose to obtain K_f . For a new flexhose assy., the user is required to employ experimental values obtained from a force-deflection test. For those flexhoses where a force-deflection test is not obtainable (i.e. a flexhose already installed permanently in a line assy.), a rough estimate of K_f may be made from the following expression:

$$K_{f} = D_{m}EN_{p}(t/h)^{3}$$
 (12)

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STRUCTURAL MODE FOR FLEXHOSE - With the bellows pressurized, longitudinal movement of the crown is restricted due to the braid. The root may move in-phase or out-of-phase with the adjacent convolute with a single degree of freedom. Therefore, the number of masses is one; which implies $N_{\text{C}}\!=\!1$; and $k\!=\!2K_{\text{f}}$.

Figure 4. Lumped Spring-Mass Mechanical Model for Flexhose

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- Step B. Calculate the elemental metal mass using Equ. (3) as done previously.
- Step C. Calculate the in-phase and out-of-phase longitudinal mode elemental fluid masses, $m_{\hbox{\footnotesize{IP}}}$ & $m_{\hbox{\footnotesize{OP}}},$ respectively from the expressions:

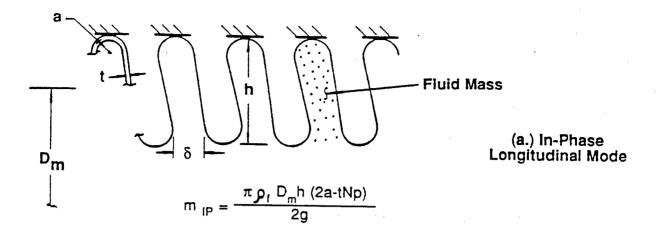
$$m_{IP} = \frac{\mathcal{H} \mathcal{P}_{f} D_{m} h (2a-tNp)}{2g}$$
 (13)

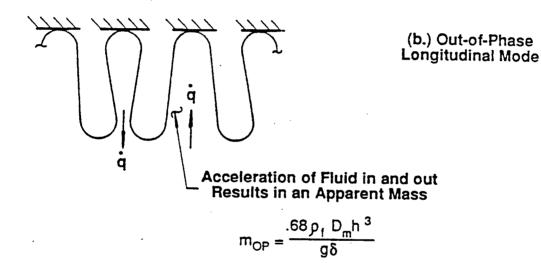
$$m_{OP} = \frac{0.68 p_f D_m h^3}{9 \delta} \tag{14}$$

NOTE: For a flexhose there are, so far as is presently known, only three possible structural vibration modes which may be flow-excited. They are the in-phase and out-of-phase longitudinal modes, and the local convolute bending mode as illustrated in Fig. 5. This is true only if the braid is maintained in full contact with all convolute crowns. Should this not be the case, the engineer is cautioned that some sections of the hose may behave as free bellows and should be treated accordingly.

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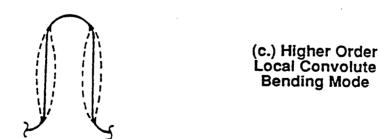


Figure 5. Summary of Flexible Hose Vibration Modes

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Step D. Calculate the in-phase, out-of-phase, and convolute bending mode frequencies from the respective expressions:

$$f_{IP} = \frac{1}{2\pi} \sqrt{\frac{2k}{m_m + m_{IP}}}$$
 (15)

$$f_{OP} = \frac{1}{2\pi} \sqrt{\frac{2k}{m_{\rm m} + m_{OP}}}$$
 (16)

$$f_{CB} = \frac{1}{2\mathcal{H}} \sqrt{\frac{8k}{m_{m} + m_{OP}}}$$
 (17)

As of now, there is no provision in the computer program for flexhose calculations. These must be done by hand.

2.2 Flow Excitation Range Calculation

Each bellows mode and flexhose mode may experience flow excitation over a fluid velocity range from a lower limit ($V_{\mbox{low}}$) to an upper limit ($V_{\mbox{up}}$) defined as

$$V_{low} = \frac{f(N)\sigma}{S_{\sigma u}}$$
 (18)

$$v_{up} = \frac{f(N)\sigma}{S_{\sigma 1}}$$
 (19)

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where σ = inside convolute width = 2a + tN_p

 $S_{\sigma u}$ = upper limit Strouhal number

 $S_{\sigma 1}$ = lower limit Strouhal number

f(N) = each modal frequency calculated for a
free bellows or flexhose

For a free bellows: Those longitudinal mode frequencies given in equ. (9) and the convolute bending mode frequency given in equ. (10).

For a flexhose: The in-phase, out-of-phase, and convolute bending mode frequencies given in equs. (15), (16), and (17) respectively.

It has been found that for most bellows and flexhose configurations, $S_{\sigma u} = 0.3$ and $S_{\sigma l} = 0.1$. The optimum or most severe excitation for each bellows and flexhose mode will occur at a critical velocity (V*) related to the critical Strouhal number ($S_{\sigma c}$) as follows:

$$V^* = \frac{f(N)\sigma}{S_{\sigma C}}$$
 (20)

where $S_{C} = 0.2$

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The possible flow excitation range of bellows and flexhoses may be predicted as follows:

- (a) Calculate the lowest and highest bellows and flexhose excitation frequency for all longitudinal modes and the convolute bending mode as summarized in paragraphs 2.1.1 and 2.1.2.
- (b) Calculate the limits of fluid velocity ($V_{\mbox{low}}$ and $V_{\mbox{up}}$) corresponding to these two frequencies.
- (c) Compare this flow-induced velocity range with the known operating range of the bellows. If an overlap of these ranges exist, then excitation may occur.

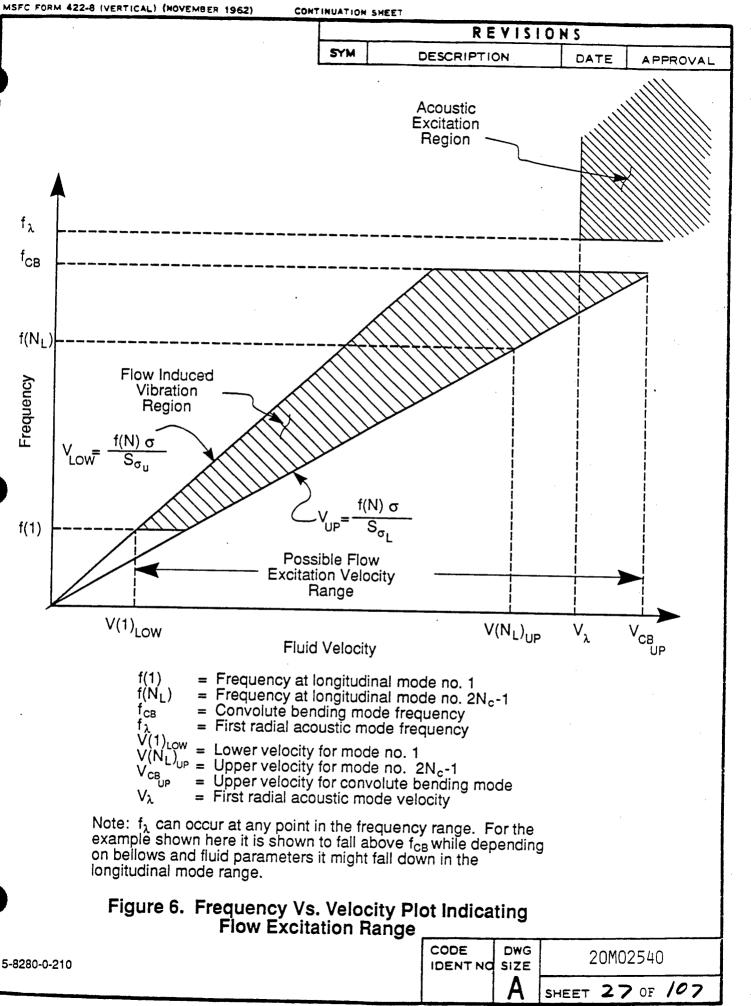
A graphical illustration of predicting the possible flow excitation range is given in Figure 6.

2.3 First Radial Acoustic Mode Resonance Calculation
(Gas Medium Only)

For a bellows or flexhose whose internal flow medium is a gas, there can occur a radial acoustic resonance. This acoustic mode can occur in addition to the longitudinal modes

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and convolute bending mode. The frequency at which the first radial acoustic mode occurs is given by:

$$f_{\lambda} = \frac{(FNCO)(C_{\phi})}{2\%r_{i}}$$
 (21)

where C_ϕ is the speed of sound and is determined from fluid property data or if the fluid behaves approximately as an ideal gas then one can use the equation below.

$$C_{\phi} = \sqrt{\frac{\chi(P+14.7)g}{P_{f}}}$$

P = fluid pressure

FNCO =
$$3.8-16.72 (h/r_i)^2 + 13.67 (h/r_i)^3$$
 for $0 \le (h/r_i) < 0.4$
FNCO = $-.336+.935 (h/r_i)^{-1}$ for $0.4 \le (h/r_i) \le 1.0$

where
$$r_i = D_i/2$$

If the longitudinal mode or convolute bending mode frequencies are greater than or equal to the first radial acoustic mode frequency, then the flow-induced stress value (FIS) is multiplied by an acoustic factor of five (5) to account for acoustic amplification. Calculation of FIS is

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described in paragraph 2.4. If the longitudinal mode or convolute bending mode frequencies are less than the first radial acoustic mode frequency, then FIS is not adjusted by an acoustic factor.

The velocity at which the first radial acoustic mode occurs is given by

$$V_{\lambda} = \frac{f_{\lambda} \sigma}{s_{\sigma c}}$$

where $S_{\sigma'C} = 0.2$

2.4 <u>Flow-Induced Stress Calculation</u>

In all velocity range overlap situations, flow-induced vibrations must be assumed to exist. Therefore, flow-induced stresses must be calculated in order to determine if a given bellows or flexhose configuration meets the design criteria of infinite life (see paragraph 1.4). Flow-induced stresses can be calculated from the following equation:

FIS = (EE)
$$\left(\frac{c^* t P_D}{v' SSR \delta}\right)$$
 (E) $(C_{NP}) (C_E) \left(\frac{1}{N_P}\right)$ (22)
where EE = 1 + 0.1 $\left(\frac{400}{SSR}\right)^2$ (NON-DIM)

NOTE: The number 400 in the above equation is a reference specific spring rate having the units lbf/in^2 .

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For a bellows:
$$SSR = \frac{K_a N_C}{D_m N_p}$$

For a flexhose:
$$SSR = \frac{K_f N_c}{D_m N_p}$$
 and $N_c=1$

For all modes except the convolute bending mode use C^{*} equation below.

$$C^* = \frac{c_1}{c_2 + (V')^2} + \frac{c_3|\sin(180V')|}{c_4 + (V')^2} + c_5$$
 (NON-DIM)

For the convolute bending mode use $C^* = 0.4$

$$V' = V^*$$

where for a bellows:

 V_C = the critical velocity for longitudinal mode $N=N_C$

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where for a flexhose:

$$V_{C}$$
 = the critical velocity for the flexhose out-of-phase mode with frequency f_{OP} given in equ. (16), so that

$$V_{C} = \frac{f_{OP} \sigma}{S_{\sigma C}}$$
where $S_{\sigma C} = 0.2$

$$P_{D} = \frac{\mathcal{P}_{f}(V^{*})^{2}}{2g}$$

$$\delta = \lambda - \sigma$$

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$$C_{NP} = \begin{cases} 1.0 & \text{for } N_p = 1 \\ 1.0 - \frac{C_6(\sigma/h)}{1.0 + C_7(V')^2} & \text{for } N_p = 2,3,... \end{cases}$$

$$C_{\rm E} = \begin{cases} 1.0 & \text{For no elbow present} \\ & \text{upstream of bellows.} \end{cases}$$

$$1.0 + \frac{4.7}{2.0 + {\rm L/D}} & \text{For elbow present} \\ & \text{upstream of bellows.} \end{cases}$$

where L = distance from elbow termination to the first bellows convolute

D = inside pipe diameter

The coefficients C_1 , C_2 ,..., C_7 are non-dimensional empirical coefficients derived from the test data in reference 1 and have the following values:

$$C_1 = 0.13$$
 $C_2 = 0.462$
 $C_3 = 1.0$
 $C_4 = 10.0$
 $C_5 = 0.06$
 $C_6 = 1.25$
 $C_7 = 5.5$

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2.4.1 <u>Uncertainty Factors</u>

A uncertainty factor (UF) is to be applied to the basic theoretical value of flow-induced stress (FIS) resulting in a corrected stress value given by

$$FISC = (FIS)(UF)$$
 (23)

where UF is determined as follows:

- (a) For a free bellows if measured spring rate is used, UF \geq 1.5
- (b) For a free bellows where spring rate is estimated from equ. (2), UF \geq 2.0
- (c) For a flexhose where measured spring rate is used, UF > 2.0
- (d) For a flexhose where spring rate is estimated from equ. (12), UF \geq 2.5
- (e) For a bellows or flexhose with radial acoustic resonance, multiply the above factors (a) through (d) by 1.5.

Note that these uncertainty factors are applied only to account for uncertainties in the analysis and data base and should not be confused with programmatic safety factors.

In addition to the above uncertainty factors, (a) through (e), an acoustic factor is to be applied in the case

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of radial acoustic resonance, as discussed already in paragraph 2.3. If any modal frequency for a given bellows or flexhose mode is greater than or equal to the first radial acoustic mode frequency (f_2), then an acoustic factor of five (5) must also be applied to the flow-induced stress value (FIS). If the modal frequency is less than the first radial acoustic mode frequency, then FIS is not adjusted by an acoustic factor.

2.5 <u>Bellows</u> and <u>Flexhose</u> <u>Fatigue</u> <u>Assessment</u>

After calculating the flow-induced stresses and applying the appropriate uncertainty factors, an assessment of the fatigue life must now be made. The fatigue life criterion is that the flexible line shall be designed to meet a theoretically infinite life for flow-induced vibration loads at its expected operating conditions (para. 1.4). For the assessment a comparison should be made between FISC and the endurance limit of the flexible line material.

NOTE: For a material not exhibiting a true endurance limit, an endurance limit as defined by each program shall be used.

If FISC is less than the endurance limit then infinite life is achieved. If FISC is greater than the endurance limit

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then infinite life cannot be met. This fatigue assessment assumes the mean stress of the flexible line is zero or negligible. A system level analysis to predict the overall life of the flexible line must be performed, although not required by this document. This system level analysis is discussed in paragraph 3.0.

If infinite life cannot be achieved through this fatigue assessment, then a redesign and reanalysis is necessary or the maximum flow velocity must be restricted in order to meet infinite life. If it is determined that a bellows redesign is necessary and if no other geometrical configurations are available or possible, then bellows liners may be required. Liners isolate the convolutes from flow impingement, thereby eliminating flow-induced vibration occurence. However, a weight and cost penalty may be associated with the installation of liners. Liners should be designed to minimize pressure differential (delta P), and where there is reverse flow, two-piece liners should be used.

2.6 Bellows and Flexhose Operating Velocity Assessment
Once it has been established that the bellows and
flexhose will have infinite life, it is still necessary to
limit the maximum operating velocity of the line. This is
necessary because of the uncertanties beyond the last mode

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predicted. Depending on the case, the maximum operating velocity of the bellows and flexhose shall be limited as follows:

Case A: For liquid flow in a bellows and flexhose, where infinite life is predicted for all longitudinal modes and the convolute bending mode, the maximum operating velocity shall be limited to the upper limit $(V_{\rm up})$ of the convolute bending mode.

Case B: For gas flow in a bellows and flexhose, where infinite life is predicted for all modes (longitudinal and convolute bending) and the first radial acoustic mode velocity (V_{λ}) is less than the upper limit (V_{up}) of the convolute bending mode, the maximum operating velocity shall be limited to (V_{up}) of the convolute bending mode.

Case C: If in Case B, the first radial acoustic mode velocity (V_2) was greater than the upper limit $(V_{\rm up})$ of the convolute bending mode, then the maximum operating velocity shall be limited to the lesser of $V_{\rm up}$ of the convolute bending mode or 80% of V_2 .

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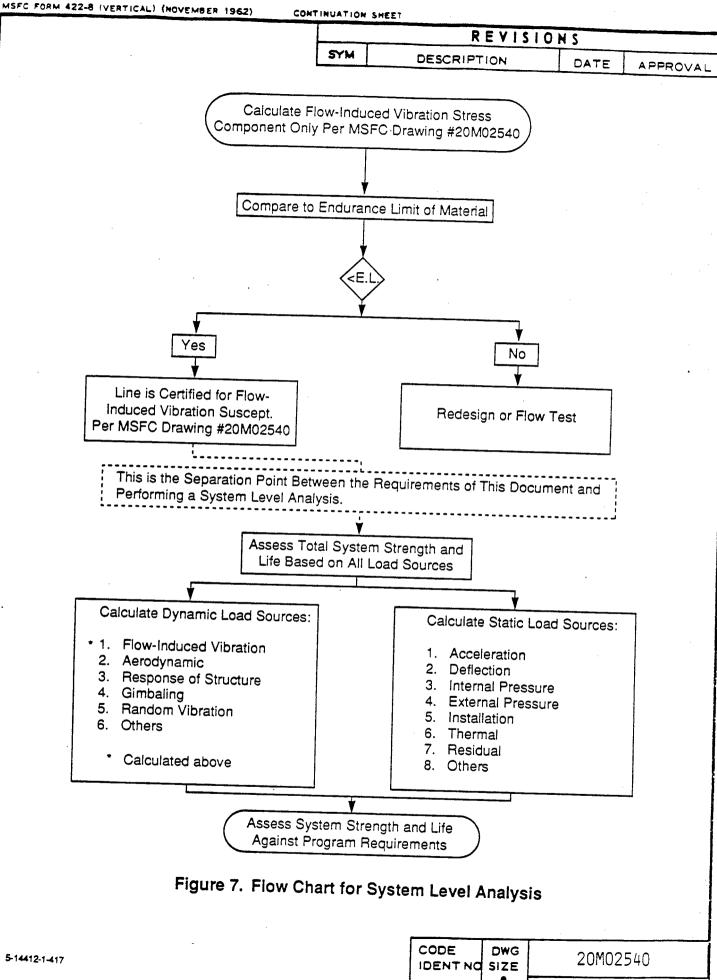
Case D: For liquid or gas flows in a bellows and flexhose, where infinite life cannot be met for all of its modes, the maximum operating velocity shall be limited to less than the lower limit (V_{low}) of the mode that first indicates finite life.

3.0 SYSTEM LEVEL ANALYSIS OF A FLEXIBLE LINE

Although the requirements of this document deal strictly with assessing the fatigue life of flexible lines from flow-induced vibration loading, the designer must realize that flow-induced vibration is only one source of a wide spectrum of loads imposed on a flexible line. Strength and fatigue analyses which include all of the load sources imposed on a flexible line must be performed. The flowchart in Figure 7 shows a general path one might follow, and some of the load sources which must be considered, in performing the analyses of a flexible line.

This section presents a few comments, guidelines, and recommendations on performing a system level analysis of a flexible line. Even though a system level analysis is not within the scope of this document and also not a requirement of this document, it should be performed sometime in the

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design process. All flexible line analysis should be reviewed and accepted by the governing agency.

- 1. Equations for the static stresses resulting from deflection and static pressure in a flexible line operating in the elastic material range may be found in references 2, 3, 4, and 5.
- 2. Flexible lines, by definition, must be flexible and capable of accommodating deflections across its length.

 Because of this, many flexible lines operate in the non-linear (plastic) material range. This phenomenon increases the difficulty of the analysis and requires that good engineering judgement and analysis procedures be used to assure that all loading effects are accounted for. The following are some items to be considered in performing a plastic analysis:
- A. A computer code capable of including large displacement and non-linear material effects will be needed. This is necessary to determine the stress field along the meridian and through the wall thickness of the flexible line. Plastic behavior of the material would be expected in the inner radius of the convolutes at the root and crown.

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- B. Gross yielding of the material should be avoided. Yielding should not occur more than 25% of the way through the wall thickness.
- C. Margins of safety may need to be calculated based on strain capability. Maximum strains must include the effects of all loading, both static and dynamic.
- Strains induced by static loading; e.g., D. deflections, pressures, etc., may be calculated by use of finite difference or finite element computer codes capable of handling large deflection and non-linear material effects. Dynamic loading effects; e.g., flow-induced vibration stresses, however, cannot be modeled directly with these methods and are usually known as loads across the flexible line or stresses in the flexible line. Several effects must be taken into consideration when calculating strains from these loads and stresses. 1. The spring rate of the flexible line changes as the line is deflected. The deflection due to the dynamic load should be calculated using the spring rate consistent with the configuration of the line when the loading is applied. The resultant strains can then be calculated using the methods employed for the static loading. 2. material modulus of elasticity will vary along the meridian

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and through the wall thickness of the flexible line, depending on if the material at that point has yielded, and the amount of yielding experienced. Therefore, the modulus of elasticity needs to be adjusted. This must be considered when calculating strains due to dynamic stresses. One factor which may be used to adjust the elastic modulus is the ratio of the spring rate of the flexible line consistent with the configuration of the line when the stresses occur, versus the spring rate of the flexible line in the undeflected condition.

- E. The change in material modulus when yielding occurs will affect the response of the flexible line to the flow. The calculation of flow-induced stresses should be repeated with an adjusted modulus if the line is found to have yielded. The method previously discussed can be used for adjusting the modulus of elasticity.
- 3. Fatigue analysis of the line must include effects of creep, low cycle, and high cycle loading. Life fractions for the creep, low cycle, and high cycle fatigue must include the required life factors and are additive. Their sum must be less than or equal to one.

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These are just a few of the issues which must be considered when performing a system level analysis of a flexible line. The analyst should assure that all loading conditions are accounted for in the analysis. The analysis should also be accompanied by a test program which simulates the operating conditions of the flexible line as closely as possible. The analysis and test program should be approved by the governing agency of the project.

4.0 FLOW TESTING

When flow testing of a bellows or flexhose is necessary, it shall be conducted in accordance with MSFC-SPEC-626 and must demonstrate a safety factor of four (4) on life.

5.0 COMPUTER PROGRAM

A computer program for conducting flow-induced vibration analysis of only a free bellows is included in Appendix C. This program calculates the frequency, flow excitation range, and flow-induced stresses for each longitudinal mode and the convolute bending mode for a free bellows. This program applies for both liquid and gas flows through the bellows. This program also calculates the first

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radial acoustic mode resonance for gas flows. The program does not conduct flow-induced vibration analysis for a flexhose and does not calculate any static stresses. These flexhose and static stress analyses have to be done by hand.

Also given in Appendix C is the input data file format along with two examples. The corresponding output files for the two examples are also given. These two examples are the same as those presented in Appendix B.

6.0 EXAMPLE PROBLEMS

There are three examples of hand calculations for flow-induced stresses given in Appendix B.

The three examples are:

Example 1.1 - - - Liquid flow through a bellows

Example 1.2 - - - Gas flow through a bellows

Example 2.0 - - - Gas flow through a flexhose

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- Salzmann, F.: "Ueber die Nachgiebigkeit von Wellrohrexpansionen," Schweiz. Bauztg., Bank 127, NR. 11, March 16, 1946, pp. 127-130.
- 3. Kellogg, M. W. Co.: "Design of Piping Systems," Second edition 1956, John Wiley and Sons, Inc., p. 222.
- 4. Anderson, W.F., 1964, <u>Analysis of Stresses in Bellows</u>, <u>Part I and Part II Design Criteria and Test Results</u>, Atomics International, Report No. NAA-SR-4527, Canoga Park, California.
- 5. <u>Standards of the Expansion Joint Manufacturers</u>
 <u>Association, Inc.</u>, 5th Edition, White Plains,
 New York, 1980.
- 6. MSFC-SPEC-626, <u>Test Control Document for Assessment of Flexible Lines for Flow-Induced Vibration.</u>

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APPENDIX A

SYMBOLS AND DEFINITIONS

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SYMBOLS

a	= Mean convolute radius = (o-tNp)/2
$\mathtt{B}_{\mathbf{N}}$	= Dimensionless frequency factor
cĒ	= Elbow factor (non-dim)
$c_{ m NP}$	<pre>= Damping modifier coefficient (non-dim)</pre>
C*	= Force and damping coefficient (non-dim)
Cø	= Speed of sound
·Di	= Inside diameter of flexible line
$D_{\mathbf{m}}$	= Mean diameter of flexible line = $(D_i + D_o)/2$
Do	= Outside diameter of flexible line
E	= Young's modulus of elasticity
fa	= First radial acoustic mode frequency
f(N)	= Modal frequency
$\mathtt{f}_{\mathtt{IP}}$	= Flexhose in-phase longitudinal mode frequency
${ t f}_{ t OP}$	= Flexhose out-of-phase longitudinal mode
	frequency
${ t f}_{ t CB}$	= Convolute bending mode frequency
f _c	= Critical frequency for mode N=N _C
fo	= Reference frequency
FNCO	= First radial acoustic mode frequency number
	(non-dim)
FIS	= Flow-induced stress
FISC	= Flow-induced stress with uncertainty factor

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SYMBOLS (Cont.)

g = Gravitational accelera	ation
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h = Mean inside convolute height = $[(D_0-D_1)/2]-tN_p$

k = Elemental spring rate of one-half of a convolution

 K_a = Overall bellows spring rate

 K_{f} = Flexhose spring rate for one complete convolution

m = Total elemental mass

 m_{m} = Elemental metal mass

 m_f = Total elemental fluid added mass

 m_{f1} = Fluid added mass

 m_{f2} = Fluid added mass

m_{IP} = Flexhose in-phase elemental fluid mass

 m_{OP} = Flexhose out-of-phase elemental fluid mass

 $N = Mode number (1,2,3...2N_C-1)$

N_C = Number of convolutes counted from the outside

 N_p = Number of plys

 P_D = Free stream dynamic pressure

P = Fluid pressure

 r_i = Inside flexible line radius = $D_i/2$

 S_{σ_1} = Lower Strouhal number (non-dim)

 S_{σ_1} = Upper Strouhal number (non-dim)

 S_{σ_C} = Critical Strouhal number (non-dim)

SSR = Specific spring rate

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SYMBOLS (Cont.)

= Endurance limit of flexible line material SEL t = Ply thickness UF = Uncertainty factor = Lower limit velocity for mode N Vlow v* = Critical velocity for mode N v_{up} = Upper limit velocity for mode N = Critical velocity for mode N=N_C Vc v' = Normalized velocity parameter = V*/Vc (non-dim) = First radial acoustic mode velocity **V**2_ X = Specific heat ratio for the gas = C_p/C_V (non-dim) 9 = Inside convolute width = 2a + tNp δ = Inside convolute gap = λ - σ 2 = Inside convolute pitch $\mathcal{P}_{\mathbf{f}}$ = Weight density of fluid (lbf/in³)

= Weight density of flexible line matl. (lbf/in³)

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<u>DEFINITIONS</u>

- Angulation Angular deflection imposed on a flexible line.
- Axial deflection Elongation or compression of a flexible line along its longitudinal axis.
- Flexhose A flexible metal hose where convolutes are partially restrained at the crown by wire braid.
- Flexible line Metal bellows or flexhose assembly that
 joins two duct sessions and permits
 relative motion between the ducts in one
 or more planes.
- Free bellows Where convolutes have unrestricted movement when exposed to fluid flow impingement.

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APPENDIX B

FLOW-INDUCED VIBRATION EXAMPLE PROBLEMS

FOR BELLOWS AND FLEXHOSE

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1.0 BELLOWS EXAMPLE PROBLEMS

1.1 <u>Liquid Medium Example</u>

Given: H_2O flowing through a 3 inch 321 stainless steel bellows at 68 ^{O}F and at 35 psig with an elbow 4 inches from the first convolute.

BELLOWS PARAMETERS

Inside convolute width, $\sigma = 0.095$ in.

Inside convolute pitch, $\lambda = 0.148$ in.

Mean inside convolute height, h = 0.325 in.

Ply thickness, t = 0.007 in.

Inside diameter, $D_i = 3.00$ in.

Outside diameter, $D_0 = 3.69$ in.

Number of convolutes, $N_C = 16$

Number of plys, $N_p = 3$

Young's modulus, E = 29.0E+06 psi

Material weight density, $p_m = 0.286$ lbf/cu. in.

Problem: Assess the fatigue life from flow-induced vibration loads for the first longitudinal mode N=1 and the longitudinal mode N= $N_{\rm C}$.

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CALCULATION PROCEDURE:

I. Frequency Calculation for Longitudinal Mode N=1.

1.
$$K_a = D_m E(N_p/N_c) (t/h)^3$$
 $K_a = 3.345(29E+06)(3/16)(.007/.325)^3$
 $K_a = 181.735 \text{ lbf/in}$
 $k = 2N_c K_a = 2(16)(181.735) = 5815.52 \text{ lbf/in}$

$$2. m = m_m + m_f$$

$$m_{m} = \frac{\gamma p_{m} t N_{p} D_{m} [\gamma a + h-2a]}{g}$$

$$a = (\sigma - tN_p)/2 = [.095 - .007(3)]/2 = 0.037 in.$$

$$m_{\rm m} = \frac{\% (.286) (.007) (3) (3.345) [\% (.037) + .325 - 2(.037)]}{32.174}$$

$$m_{\rm m}$$
 = 7.20E-04 slugs

$$m_f = K_1 m_{f1} + K_2 m_{f2} (N/N_c)$$

$$m_{f1} = \frac{\gamma \mathcal{P}_{f} D_{m} h (2a-t N_{p})}{2g}$$

$$m_{e_3} = \frac{\mathcal{H}(62.4/1728)(3.345)(.325)[2(.037)-.007(3)]}{2(32.174)}$$

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$$m_{fl} = 1.02E-04$$
 slugs

$$m_{f2} = \frac{p_f D_m h^3}{g \delta} = \frac{(62.4/1728)(3.345)(.325)^3}{32.174(.148-.095)}$$

$$m_{f2} = 2.43E-03$$
 slugs

$$m_f = 1.0(1.02E-04) + 0.68(2.43E-03)(1/16)$$

$$m_{\hat{f}} = 2.05E-04 \text{ slugs}$$

$$m = m_m + m_f = 7.20E-04 + 2.05E-04$$

$$m = 9.25E-04$$
 slugs

*3.
$$B_N = \{2[1+\cos(180(2N_C-N)/2N_C)]\}^{1/2}$$

for N=1,
$$B_1 = \{2[1+\cos(180(32-1)/32)]\}^{1/2}$$

$$B_1 = 0.0981$$

4.
$$f_0 = \frac{1}{2\pi} \sqrt{\frac{k}{m}} = \frac{1}{2\pi} \sqrt{\frac{12(5815.52)}{9.25E-04}} = 1382.40 \text{ Hz}$$

5.
$$f(1) = (f_0)(B_1)$$

 $f(1) = (1382.40)(.0981) = 135.61 Hz$

*If calculating cos in degrees, use 180; if calculating cos in radians, use γ .

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II. Velocity Range Calculation for Longitudinal Mode N=1.

$$V(N,i) = f(N)\sigma$$
 $(S_{\sigma i})$

where
$$N = 1, 2, 3..., 2N_C-1$$

 $i = 1, 2, 3$

$$S_{\sigma 1} = S_{\sigma u} = 0.3$$

$$S_{\sigma_2} = S_{\sigma_C} = 0.2$$

$$S_{03} = S_{01} = 0.1$$

$$V_{low} = V(1,1) = \frac{f(1) \sigma}{(S_{u})} = \frac{135.61(.095)}{12(0.3)} = 3.58 \text{ fps}$$

$$V^* = V(1,2) = \frac{f(1)\sigma'}{(S_{\sigma C})} = \frac{135.61(.095)}{12(0.2)} = 5.37 \text{ fps}$$

$$V_{up} = V(1,3) = f(1)\sigma' = \frac{135.61(.095)}{(S_{\sigma'1})} = 10.74 \text{ fps}$$

Flow-Induced Stress Calculation for Longitudinal Mode N=1 III.

Critical frequency (f_c) at N=N_c

$$m_f = 1.0(1.02E-04) + 0.68(2.43E-03)(16/16)$$

$$m_f = 1.75E-03$$
 slugs

$$m = m_m + m_f = 7.20E-04 + 1.75E-03$$

$$m = 2.47E-03$$
 slugs

$$0 N = N_C, B_{16} = \sqrt{2^3}$$

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$$f_{c} = (f_{o}) (B_{16})$$

$$f_{c} = \frac{1}{2 \gamma} \sqrt{\frac{12(5815.52)}{2.47E-03}} \left(\sqrt{2}\right) = 1196.4 \text{ Hz}$$

2. Critical velocity (
$$V_c$$
) at $N=N_c$

$$V_c = \frac{f_c \sigma}{(S_{\sigma c})} = \frac{1196.4(.095)}{12(0.2)} = 47.36 \text{ fps}$$

3.
$$V' = V^*/V_C = \frac{5.37}{47.36} = 0.113$$

4. SSR =
$$\frac{K_a N_c}{D_m N_p}$$
 = $\frac{181.735(16)}{3.345(3)}$ = 289.762 lbf/in²

*5.
$$C_{NP} = 1.0 - \frac{C_6 (\sigma/h)}{1 + C_7 (V')^2}$$

$$C_{NP} = 1.0 - \frac{1.25(.095/.325)}{1 + 5.5(.113)^2} = .659$$

$$c^* = \frac{c_1}{c_2 + (V')^2} + \frac{c_3|\sin(180V')|}{c_4 + (V')^2} + c_5$$

$$c^* = \frac{0.13}{0.462 + (.113)^2} + \frac{1.0|\sin[180(.113)]|}{10.0 + (.113)^2} + 0.06$$

$$C^* = 0.369$$

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6.
$$P_{D} = p_{f}(V^{*})^{2} = \frac{(62.4/1728)(5.37)^{2}(12)}{2(32.174)} = 0.194 \text{ psi}$$

7.
$$DD = \frac{C^* tP_D}{V' SSR \delta} = \frac{0.369(.007)(.194)}{0.113(289.762)(.053)} = 2.89E-04$$

*If calculating sin in degrees, use 180; if calculating sin in radians, use \mathcal{X} .

8. EE = 1.0 + 0.1
$$\left(\frac{400}{SSR}\right)^2$$
 = 1.0 + 0.1 $\left(\frac{400}{289.762}\right)^2$

$$EE = 1.191$$

9.
$$C_E = 1.0 + \frac{4.7}{2 + L/D} = 1.0 + \frac{4.7}{2 + 4/3} = 2.41$$

10. FIS = (EE) (DD) (E) (
$$C_{NP}$$
) (C_{E})

11. FIS =
$$\frac{1.191(2.89E-04)(29E+06)(.659)(2.41)}{3}$$

FIS = 5,284 psi for longitudinal mode N=1

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Following the same procedure as above the value of FIS for the $N=N_{\rm C}$ mode is determined to be

FIS = 26,873 psi for longitudinal mode $N=N_{\rm C}$

12. Uncertainty Factor:

The spring rate was estimated using equ. (2) therefore, the uncertainty factor is

UF = 2.0

and the corrected predicted flow-induced stress for longitudinal modes N=1 and N=N $_{\mbox{\scriptsize C}}$ is:

FISC = (UF)(FIS)

FISC = (2.0)(5,284) = 10,568 psi for long. mode N=1

FISC = (2.0)(26,873) = 53,747 psi for long. mode $N=N_{C}$.

IV. Fatigue Assessment

From the results above

FISC = 10,568 psi for longitudinal mode N=1

FISC = 53,747 psi for longitudinal mode $N=N_{C}$

For 321 steel at $68^{\circ}F$ the endurance limit is $S_{\rm EL}$ = 26,500 psi

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Conclusion:

For longitudinal mode N=1 FISC is less than S_{EL} , therefore, infinite life is predicted. For longitudinal mode N= \acute{N}_C FISC is greater than S_{EL} , therefore, finite life is indicated and the bellows must be redesigned if operated to a velocity capable of exciting this mode. Repeating this analysis for each mode lower than N= N_C could be performed to find the mode and corresponding maximum velocity the bellows can be operated and still achieve infinite life.

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1.2 <u>Gaseous Medium Example</u>

Given: GN_2 flowing through an 8 inch 21-6-9 steel bellows at -200 $^{
m O}F$ and at 39.3 psig with no elbow upstream.

BELLOWS PARAMETERS

Inside convolute width, $\sigma = 0.400$ in.

Inside convolute pitch, $\lambda = 0.726$ in.

Mean inside convolute height, h = 1.25 in.

Ply thickness, t = .037 in.

Inside diameter, $D_i = 8.00$ in.

Outside diameter, $D_0 = 10.574$ in.

Number of convolutes, $N_C = 7$

Number of plys, $N_D = 1$

Young's modulus, E = 28.5E+06 psi

Material weight density, $p_m = 0.282$ lbf/cu. in.

Problem: Determine the maximum safe flow velocity

which will result in a predicted infinite life

from flow-induced vibration loads.

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CALCULATION PROCEDURE:

I. Frequency Calculation for Longitudinal Mode N=1.

1.
$$K_a = D_m E(N_p/N_c) (t/h)^3$$

$$K_a = \left(\frac{8.00 + 23.574}{2}\right) \left(28.5E + 06\right) \left(\frac{1}{7}\right) \left(\frac{.037}{1.25}\right)^3$$

$$K_a = 980.61 lbf/in$$

$$k = 2N_cK_a = 2(7)(980.61) = 13,728.54 lbf/in$$

$$2. m = m_m + m_f$$

$$m_{m} = \frac{\mathcal{N} p_{m} t N_{p} D_{m} [\mathcal{A}a + h-2a]}{g}$$

$$a = (\sigma - tN_p)/2 = [.400 - .037(1)]/2 = 0.182 in$$

$$m_{m} = \gamma(.282)(.037)(1)(9.287)[\gamma(.182)+1.25-2(.182)]$$

$$32.174$$

$$m_{\rm m} = 137.93E-04$$
 slugs

$$m_f = K_1 m_{f1} + K_2 m_{f2} (N/N_c)$$

$$m_{f1} = \frac{\gamma \mathcal{P}_{f} D_{m} h (2a-t N_{p})}{2g}$$

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where
$$\mathcal{P}_{f} = \mathcal{P}_{ref}(P/P_{ref})(T_{ref}/T)$$
 (Z_{ref}/Z)

P = gas pressure, psia

P_{ref} = reference pressure = 14.7 psia

T = gas temperature, OR

Z = gas compressibility factor (non-dim)

 T_{ref} = reference temperature = 528 O R

 \mathcal{P}_{ref} = reference density (lbf/ft³) = $P_{ref}/RT_{ref}Z_{ref}$ from gas law

$$\mathcal{P}_{\text{ref}} = \frac{14.7(144)}{54.92(528)(1.0)} = 0.073 \text{ lbf/ft}^3$$

$$\mathcal{P}_{f} = \left(\frac{0.073}{1728}\right) \left(\frac{39.3 + 14.7}{14.7}\right) \left(\frac{528}{-200 + 460}\right) \left(\frac{1.0}{.982}\right)$$

$$\mathcal{P}_{f} = 3.21E-04 \text{ lbf/in}^3$$

$$m_{f1} = \frac{\mathcal{M}(3.21E-04)(9.287)(1.25)[2(.182)-.037(1)]}{2(32.174)}$$

 $m_{f1} = 5.95E-05$ slugs

$$m_{f2} = \frac{\rho_f D_m h^3}{32.174(.726-.400)} = \frac{3.21E-04(9.287)(1.25)^3}{32.174(.726-.400)}$$

 $m_{f2} = 5.55E-04$ slugs

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$$m_f = 1.0(5.95E-05) + 0.68(5.55E-04)(1/7)$$

$$m_f = 1.13E-04$$
 slugs

$$m = m_m + m_f = 137.93E-04 + 1.13E-04$$

 $m = 13.91E-03 slugs$

*3.
$$B_N = \{2[1+\cos(180(2N_C-N)/2N_C)]\}^{1/2}$$

for N=1, $B_1 = \{2[1+\cos(180(14-1)/14)]\}^{1/2}$
 $B_1 = 0.2239$

4.
$$f_0 = \frac{1}{2\pi} \sqrt{\frac{k}{m}} = \frac{1}{2\pi} \sqrt{\frac{12(13728.54)}{13.91E-03}} = 547.72 \text{ Hz}$$

5.
$$f(1) = (f_0)(B_1) = 547.72(0.2239) = 122.63 Hz$$

*If calculating cos in degrees, use 180; if calculating cos in radians, use
$$\gamma$$
.

II. Velocity Range Calculation for Longitudinal Mode N=1

$$V_{low} = V(1,1) = \frac{f(1)\sigma}{S_{\sigma u}} = \frac{122.63(.400)}{12(0.3)} = 13.63 \text{ fps}$$

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$$V^* = V(1,2) = f(1)\sigma = \frac{122.63(.400)}{S_{\sigma C}} = 20.44 \text{ fps}$$

$$V_{up} = V(1,3) = \frac{f(1)\sigma'}{S_{\sigma'1}} = \frac{122.63(.400)}{12(0.1)} = 40.88 \text{ fps}$$

III. First Radial Acoustic Mode Resonance Calculation

1.
$$h/r_i = 1.25/(8.00/2) = .3125$$

2. FNCO =
$$3.8-16.72(h/r_i)^2 + 13.67(h/r_i)^3$$

3. Speed of Sound

FNCO = 2.58

$$C_{\phi} = \sqrt{\frac{\chi(P+14.7)g}{Pf}} = \sqrt{\frac{1.40(39.3+14.7)(32.174)}{3.21E-04(12)}}$$

$$C_{\phi} = 794.6 \text{ fps}$$

- 4. First Radial Acoustic Mode Frequency $f_{\lambda} = \frac{(FNCO)(C_{0})}{2\pi r_{1}} = \frac{12(2.58)(794.6)}{2\pi (4.0)} = 978.84 \text{ Hz}$
- 5. First Radial Acoustic Mode Velocity $V_{\lambda} = \frac{f_{\lambda}\sigma}{s_{\sigma}c} = \frac{978.84(.400)}{12(0.2)} = 163.14 \text{ fps}$

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IV. Flow-Induced Stress Calculation for Longitudinal Mode N=1

1. Critical frequency (
$$f_c$$
) at $N=N_c$

$$m_f = 1.0(5.95E-05) + 0.68(5.55E-04)(7/7)$$

$$m_f = 4.37E-04$$
 slugs

$$m = m_m + m_f = 137.93E-04 + 4.37E-04$$

$$m = 14.23E-03$$
 slugs

at N=N_C, B₇ =
$$\sqrt{2}$$

$$f_c = (f_0)(B_7)$$

$$f_C = \frac{1}{2\pi} \sqrt{\frac{12(13728.54)}{14.23E-03}}$$
 ($\sqrt{2}$) = 765.84 Hz

2. Critical velocity (V_c) at $N=N_c$

$$V_c = \frac{f_c \sigma}{S_{\sigma C}} = \frac{765.84(.400)}{12(0.2)} = 127.64 \text{ fps}$$

3.
$$V' = V^*/V_C = 20.44/127.64 = 0.160$$

4. SSR =
$$\frac{K_a N_c}{D_m N_D} = \frac{980.61(7)}{9.287(1)} = 739.13 \text{ lbf/in}^2$$

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5.
$$C_{NP} = 1.0 \text{ for } N_p = 1$$

*6.
$$c^* = \frac{c_1}{c_2 + (V')^2} + \frac{c_3 |\sin(180V')|}{c_4 + (V')^2} + c_5$$

$$C^* = \frac{0.13}{.462 + (.160)^2} + \frac{1.0|\sin[180(.160)]!}{10 + (.160)^2} + 0.06$$

$$C^* = 0.375$$

*If calculating sin in degrees, use 180; if calculating sin in radians, use γ .

7.
$$P_D = \frac{p_f(v^*)^2}{2g} = \frac{3.21E - 04(20.44)^2(12)}{2(32.174)} = 0.025 \text{ psi}$$

8. DD =
$$\frac{c^* tP_D}{v' \text{SSR } \delta}$$
 = $\frac{0.375(.037)(.025)}{0.160(739.13)(.326)}$ = 9.00E-06

9. EE = 1.0 + 0.1
$$\left(\frac{400}{SSR}\right)^2$$
 = 1.0 + 0.1 $\left(\frac{400}{739.13}\right)^2$ = 1.029

10. $C_E = 1.0$ for no elbow present upstream

11. FIS = (EE) (DD) (E) (
$$C_{NP}$$
) (C_E)
$$N_p$$

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FIS =
$$(1.029)(9.00E-06)(28.5E+06)(1.0)(1.0)$$

1.0

$$FIS = 263.94 psi$$

12.
$$f(1) = 122.63 \text{ Hz} < f_2 = 978.84 \text{ Hz}$$
; therefore

FIS = 263.94 psi for long. mode N=1

NOTE: If $f(N) \ge f_2$, then FIS is multiplied by an acoustic factor of five (5).

13. Uncertainty Factor:

The spring rate was estimated using equ. (2) therefore, the uncertainty factor is

$$UF = 2.0$$

and the corrected predicted flow-induced stress for longitudinal mode N=1 is:

$$FISC = (UF)(FIS)$$

FISC = (2.0)(263.94) = 527.88 psi for long. mode N=1

V. Calculations for Longitudinal Modes N=2 thru 13.

A similar proceedure has been applied to the remaining bellows longitudinal modes (N=2 thru 13) and the results are

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summarized in Table B-1. The convolute bending mode is calculated next as shown below and is also summarized in Table B-1.

VI. Frequency Calculation for Convolute Bending Mode.

$$f_{CB} = \frac{1}{2\pi} \sqrt{\frac{8k}{m_m + .68 m_{f2}}}$$

where k, m_{m} , and m_{f2} were previously calculated

$$f_{CB} = \frac{1}{2\pi} \sqrt{\frac{8(13,728.54)(12)}{137.93E-04 + .68(5.55E-04)}}$$

$$f_{CB} = 1534.89 \text{ Hz}$$

VII. Velocity Range Calculation for Convolute Bending Mode.

$$V_{low} = f_{CB} \sigma = 1534.89(.400) = 170.54 \text{ fps}$$

$$S_{\sigma_{11}}$$

$$V^* = \frac{f_{CB}\sigma}{s_{\sigma C}} = \frac{1534.89(.400)}{12(0.2)} = 255.82 \text{ fps}$$

$$v_{up} = \frac{f_{CB}\sigma}{s_{\sigma 1}} = \frac{1534.89(.400)}{12(0.1)} = 511.63 \text{ fps}$$

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VIII. Flow-Induced Stress Calculation for Convolute Bending Mode.

1.
$$V' = V^*/V_C$$
 where V_C was previously calculated

$$V' = 255.82/127.64 = 2.004$$

2.
$$C_{NP} = 1.0 \text{ for } N_p=1$$

3.
$$C^* = 0.4$$
 for the convolute bending mode

4.
$$P_D = \frac{p_f(V^*)^2}{2g} = \frac{12(3.21E-04)(255.82)^2}{2(32.174)}$$

$$P_D = 3.918 \text{ psi}$$

5. DD =
$$\frac{c^*tP_D}{v'ssr \delta}$$
 where SSR and δ were previously calculated

$$DD = \frac{(0.4)(.037)(3.918)}{(2.004)(739.13)(.326)} = 1.20E-04$$

7.
$$C_E$$
 = 1.0 for no elbow present upstream

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8. FIS =
$$\frac{\text{(EE) (DD) (E) (C}_{NP}) (C_{E})}{N_{P}}$$

FIS =
$$(1.029)(1.20E-04)(28.5E+06)(1.0)(1.0)$$

1.0

$$FIS = 3,519.18 psi$$

- 9. The convolute bending mode frequency is higher than the first radial acoustic mode frequency $f_{CB} = 1534.89 \text{ Hz} > f_2 = 978.84 \text{ Hz}$ therefore, an acoustic factor of 5 is applied to FIS below.
- 10. Uncertainty Factors:

The spring rate was estimated using equation (2) and since radial acoustic resonance is predicted, therefore, the uncertainty factor is

$$UF = (2.0)(1.5) = 3.0$$

An acoustic factor of 5 is also applied to FIS to account for an increase in stress levels due to radial acoustic resonance. The corrected

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flow-induced stress for the convolute bending mode is now:

$$FISC = (UF) (5.0) (FIS)$$

$$FISC = (3.0)(5.0)(3,519.18)$$

FISC = 52,788 psi for the convolute bending mode

IX. Fatigue Assessment

For 21-6-9 steel at -200°F the endurance limit is
$$s_{\rm EL} \, = \, 47,000 \ {\rm psi}$$

Conclusion:

From the results summarized in Table B-1 and in Figure B-1 we now compare the predicted FISC values with the $S_{\rm EL}$ value given above. If FISC is less than $S_{\rm EL}$, infinite life is predicted. If FISC is greater than $S_{\rm EL}$, the life of the bellows is finite. All 13 longitudinal modes have FISC values less than $S_{\rm EL}$. The convolute bending mode FISC value is greater than $S_{\rm EL}$ because the first radial acoustic mode resonance occurs at 163.14 fps.

Based on the above, infinite life is predicted until the convolute bending mode frequency is reached. This

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convolute bending mode has a flow excitation range from V_{low} = 170.54 fps to V_{up} = 511.63 fps with the optimum or most severe flow excitation occurring at V^* = 255.82 fps. Therefore, the maximum safe flow velocity should be limited, according to case D in paragraph 2.6, to less than 170.54 fps to maintain a predicted infinite life.

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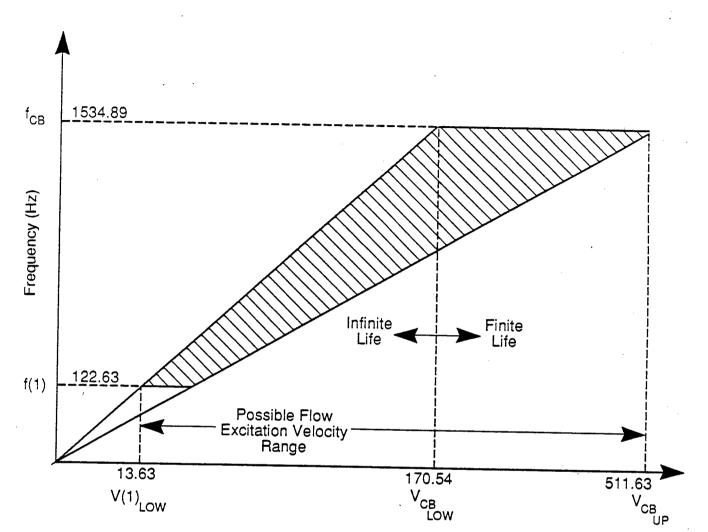
MODE	VELOCITY (ft/sec)	FREQUENCY (Hz)	ACOUSTIC FACTOR	FIS (psi)	UNCERTAINTY FACTOR	FISC (psi)	LIFE
1	20.44	122.63	1.0	263.94	2.0	527.88	Infinite
2	40.56	243.37	1.0	522	2.0	1045	Infinite
3	60.09	360.53	1.0	⁻ 719	2.0	1438	Infinite
4	78.79	472.71	1.0	824	2.0	1649	Infinite
5	96.42	578.53	1.0	835	2.0	1671	Infinite
6	112.78	676.70	1.0	768	2.0	1536	Infinite
7	127.64	765.99	1.0	654	2.0	1309	Infinite
8	140.89	845.34	1.0	804	2.0	1609	Infinite
9	152.30	913.78	1.0	937	2.0	1875	Infinite
10	161.75	970.49	1.0	1041	2.0	2082	Infinite
11	169.14	1014.82	5.0	5563	3.0	16,690	Infinite
12	174.37	1046.25	5.0	5784	3.0	17,352	Infinite
13	177.40	1064.43	5.0	5897	3.0	17,690	Infinite
Convolute Bending	255.82	1534.89	5.0	17,596	3.0	52,788	Finite
Firs	st Radial Acc	oustic Mode R	esonance:	٧ ₂ = 163.	14 fps, f ₂	= 978.84 н	z

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Table B-1. Gaseous Medium Example Summary



Fluid Velocity (FT/SEC)

Figure B-1. Frequency vs. Velocity Plot Indicating Flow Excitation Range for Example 1.2

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2.0 FLEXHOSE EXAMPLE PROBLEM

Given: Gaseous helium flowing through a flexible metal hose at 75°F and at 600 psig.

The flexhose is made of 21-6-9 steel and it is assumed the braid will be in contact with all of the convolute crowns. There is no elbow upstream of the flexhose.

FLEXHOSE PARAMETERS

Inside convolute width, $\sigma = 0.072$ in.

Inside convolute pitch, $\lambda = 0.104$ in.

Mean inside convolute height, h = 0.154 in.

Ply thickness, t = 0.010 in.

Inside diameter, $D_i = 1.850$ in.

Outside diameter, $D_0 = 2.198$ in.

* Number of convolutes, $N_C = 32$

Number of plys, $N_D = 2$

Young's modulus, E = 28.5E+06 psi

Material weight density, $p_m = 0.282$ lbf/cu. in.

* NOTE: This is the actual number of convolutes of the flexhose, however, in the analysis $N_{\rm C}$ = 1 is used.

Problem: Determine if the flexhose will have infinite life from flow-induced vibration loads when operated at 800 fps flow velocity.

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CALCULATION PROCEDURE:

I. Frequency Calculation for the Three Flexhose Modes.

1.
$$K_f = D_m E(N_p/N_c) (t/h)^3$$

$$K_f = 2.024(28.5E+06)(2/1)(.010/.154)^3$$

$$K_f = 31,588 \text{ lbf/in}$$

$$k = 2 K_f = 2(31,588) = 63,176 \text{ lbf/in}$$

2.
$$m_{m} = \frac{\pi p_{m} t N_{p} D_{m} [\pi a + h - 2a]}{g}$$

$$m_{\text{m}} = \frac{\gamma(.282)(.010)(2)(2.024)[\gamma(.026)+.154-2(.026)]}{32.174}$$

 $a = (\sigma - tN_p)/2 = [.072 - .010(2)]/2 = 0.026 in$

$$m_{\rm m} = 2.05E-04$$
 slugs

3.
$$m_{IP} = \frac{\mathcal{H}p_f D_m h (2a-tN_p)}{2g}$$

where
$$\mathcal{P}_{f} = \mathcal{P}_{ref}(P/P_{ref})(T_{ref}/T)(Z_{ref}/Z)$$

These terms are explained in example 1.2

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$$\mathcal{P}_{\text{ref}} = \frac{14.7(144)}{385.96(528)(1.0)} = 0.010 \text{ lbf/ft}^3$$

$$\mathcal{P}_{f} = \left(\frac{.010}{1728}\right) \left(\frac{600+14.7}{14.7}\right) \left(\frac{528}{535}\right) \left(\frac{1.0}{1.02}\right)$$

$$\mathcal{P}_{f} = 2.34E-04 \text{ lbf/in}^3$$

$$m_{\text{IP}} = \frac{\gamma(2.34E-04)(2.024)(.154)[2(.026)-.010(2)]}{2(32.174)}$$

$$m_{IP} = 1.13E-07$$
 slugs

$$m_{OP} = \frac{0.68 \mathcal{P}_f D_m h^3}{g \delta}$$

$$m_{OP} = \frac{0.68(2.34E-04)(2.024)(.154)^3}{32.174(.032)}$$

$$m_{OP} = 1.14E-06$$
 slugs

5.
$$f_{IP} = \frac{1}{2\pi} \sqrt{\frac{2k}{m_m + m_{IP}}}$$

$$f_{IP} = \frac{1}{27} \sqrt{\frac{2(63,176)12}{2.05E-04+1.13E-07}}$$

$$f_{TP} = 13,684 \text{ Hz}$$
 (In-Phase Mode)

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6.
$$f_{OP} = \frac{1}{2\% \sqrt{m_m + m_{OP}}}$$

$$f_{OP} = \frac{1}{2\pi} \sqrt{\frac{2(63,176)12}{2.05E-04+1.14E-06}}$$

7.
$$f_{CB} = \frac{1}{2\pi} \sqrt{\frac{8k}{m_m + m_{OP}}}$$

$$f_{CB} = \frac{1}{2\pi} \sqrt{\frac{8(63,176)12}{2.05E-04+1.14E-06}}$$

$$f_{CB} = 27,299 \text{ Hz}$$
 (Convolute Bending Mode)

II. Velocity Range Calculation for the Three Flexhose Modes:

1. In-Phase Mode:

$$V_{low} = \frac{(f_{IP})\sigma}{s_{du}} = \frac{13,684(.072)}{12(0.3)} = 273.7 \text{ fps}$$

$$V^* = \frac{(f_{IP})\sigma}{s_{\sigma c}} = \frac{13.684(.072)}{12(0.2)} = 410.5 \text{ fps}$$

$$V_{up} = \frac{(f_{Ip})\sigma}{s_{\sigma 1}} = \frac{13,684(.072)}{12(0.1)} = 821.0 \text{ fps}$$

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2. Out-of-Phase Mode:

$$V_{low} = \frac{(f_{OP})\sigma}{s_{\sigma u}} = \frac{13,650(.072)}{12(0.3)} = 273.0 \text{ fps}$$

$$V^* = \frac{(f_{OP})\sigma}{s_{\sigma c}} = \frac{13,650(.072)}{12(0.2)} = 409.5 \text{ fps}$$

$$V_{up} = \frac{(f_{OP})\sigma}{s_{\sigma l}} = \frac{13,650(.072)}{12(0.1)} = 819.0 \text{ fps}$$

3. Convolute Bending Mode:

$$V_{low} = \frac{(f_{CB})\sigma}{S_{\sigma u}} = \frac{27,299(.072)}{12(0.3)} = 546.0 \text{ fps}$$

$$V^* = \frac{(f_{CB})\sigma}{S_{\sigma c}} = \frac{27,299(.072)}{12(0.2)} = 819.0 \text{ fps}$$

$$V_{up} = \frac{(f_{CB})\sigma}{S_{\sigma c}} = \frac{27,299(.072)}{12(0.1)} = 1637.9 \text{ fps}$$

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III. First Radial Acoustic Mode Resonance Calculation

1.
$$h/r_i = .154/(1.850/2) = .166$$

2. FNCO =
$$3.8-16.72(h/r_i)^2+13.67(h/r_i)^3$$

FNCO = 3.40

3. Speed of Sound:

$$C_{\phi} = \sqrt{\frac{\chi(P+14.7)g}{P_f}} = \sqrt{\frac{1.66(600+14.7)(32.174)}{2.34E-04(12)}}$$
 $C_{\phi} = 3419.32 \text{ fps}$

4. First Radial Acoustic Mode Frequency:

$$f_{\lambda} = \frac{(FNCO)(C_{\phi})}{2\pi r_{i}} = \frac{12(3.40)(3419.32)}{2\pi (.925)}$$

$$f_2 = 24,004 \text{ Hz}$$

5. First Radial Acoustic Mode Velocity:

$$V_{\lambda} = \frac{f_{\lambda}\sigma}{S_{\alpha C}} = \frac{24,004(.072)}{12(0.2)}$$

$$V_2 = 720.12 \text{ fps}$$

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- IV. Flow-Induced Stress Calculation for the Three Flexhose Modes
 - 1. Critical frequency (f_c)

The critical frequency corresponds to the out-of-phase mode frequency, $f_{\mbox{\scriptsize OP}}$, previously calculated.

$$f_{C} = f_{OP} = 13,650 \text{ Hz}$$

2. Critical velocity (V_c)

The critical velocity corresponds to the out-of-phase mode critical velocity, \mathbf{V}^* , previously calculated.

$$V_C = V^*$$
 (out-of-phase) = 409.5 fps

3.
$$V'_{IP} = V^*/V_{C} = 410.5/409.5 = 1.00$$

$$V'_{OP} = V^*/V_{C} = 409.5/409.5 = 1.00$$

$$V'_{CB} = V^*/V_{C} = 819.0/409.5 = 2.00$$

4.
$$SSR = \frac{K_f N_c}{D_m N_p} = \frac{31.588(1)}{2.024(2)} = 7803.4 \text{ lbf/in}^2$$

where $N_C=1$ for a flexhose

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5.
$$c_{NP} = 1.0 - \frac{c_6 (\sigma/h)}{1 + c_7 (V')^2}$$

Out-of-Phase Mode
$$C_{NP} = .91$$
 $C_{NP} = .91$

Convolute Bending Mode
$$C_{NP} = .97$$

6. C* for in-phase and out-of-phase modes

$$c^* = \frac{c_1}{c_2 + (v')^2} + \frac{c_3 |\sin(180v')|}{c_4 + (v')^2} + c_5$$

In-Phase Mode
$$C^* = .15$$

Out-of-Phase Mode
$$C^* = .15$$

For the convolute bending mode use $C^* = 0.4$

7.
$$P_D = \frac{p_f(v^*)^2}{2g}$$

In-Phase Mode
$$P_D = \frac{(2.34E-04)(410.5)^212}{2(32.174)} = 7.35 \text{ psi}$$

Out-of-Phase Mode
$$P_D = \frac{(2.34E-04)(409.5)^212}{2(32.174)} = 7.32 \text{ psi}$$

Convolute Bending Mode
$$P_D = \frac{(2.34E-04)(819.0)^212}{2(32.174)} = 29.27 \text{ psi}$$

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$$DD = \frac{C^* tP_D}{V' SSR \delta}$$

$$DD = \frac{.15(.010)(7.35)}{1.0(7803.4)(.032)} = 4.41E-05$$

9. EE = 1+0.1
$$\left(\frac{400}{SSR}\right)^2$$
 = 1+0.1 $\left(\frac{400}{7803.4}\right)^2$ = 1.00

10.
$$C_E = 1.0$$
 for no elbow present upstream

11. FIS =
$$(EE) (DD) (E) (C_{NP}) (C_E)$$

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12. Since the first radial acoustic mode frequency (f₂) is greater than the in-phase and out-of phase mode frequencies, no acoustic factor of 5 is applied to FIS. However, the convolute bending mode frequency is higher than the first radial acoustic mode frequency.

$$f_{CB} = 27,299 \text{ Hz} > f_{\lambda} = 24,004 \text{ Hz}$$

therefore, an acoustic factor of 5 is applied to FIS below.

13. Uncertainty Factor:

For the in-phase and out-of-phase modes the spring rate was estimated using equation (12) and no radial acoustic resonance is predicted, therefore the uncertainty factor is

$$UF = 2.5$$

For the convolute bending mode the spring rate was estimated using equation (12) and since radial acoustic resonance is predicted, therefore the uncertainty factor is

$$UF = (2.5)(1.5) = 3.75$$

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The corrected predicted flow-induced stress for each of the three flexhose modes are:

$$FISC = (UF)(FIS)$$

The results of the three flexhose modes are summarized in Table B-2.

V. Fatigue Assessment

For 21-6-9 steel at $75^{\circ}F$ the endurance limit is $S_{\rm EL}$ = 31,000 psi

Conclusion:

From the results summarized in Table B-2 and in Figure B-2, we now compare the predicted FISC values with the $S_{\rm EL}$ value given above. If FISC is less than $S_{\rm EL}$, infinite life is predicted. If FISC is greater than $S_{\rm EL}$ then a finite life is predicted. The in-phase and out-of-phase mode FISC values are less than $S_{\rm EL}$. The convolute

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bending mode FISC value is greater than \mathbf{S}_{EL} because radial acoustic resonance occurs at 720.12 fps.

Based on the above infinite life is predicted until the convolute bending mode frequency is reached. This convolute bending mode has a flow excitation range from $V_{low}=546.0$ fps to $V_{up}=1637.9$ fps with the optimum or most severe flow excitation occurring at $V^*=819.0$ fps. Therefore, the maximum safe flow velocity should be limited, according to case D in paragraph 2.6, to less than 546.0 fps to maintain a predicted infinite life. This flexhose will not have infinite life when operated at 800 fps flow velocity.

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MODE	CRITICAL VELOCITY ft/sec	FREQUENCY Hz	ACOUSTIC FACTOR	FIS (psi)	UNCERTAINTY FACTOR	FISC (psi)	LIFE
Out-of-Phase	409.5	13,650	1.0	570.6	2.5	1426.5	Infinite
In-Phase	410.5	13,684	1.0	571.9	2.5	1429.8	Infinite
Convolute Bending	819.0	27,299	5.0	3240.0	3.75	60,750	Finite

First Radial Acoustic Mode Resonance: $V_{\lambda} = 720.12$ fps, $f_{\lambda} = 24,004$ Hz

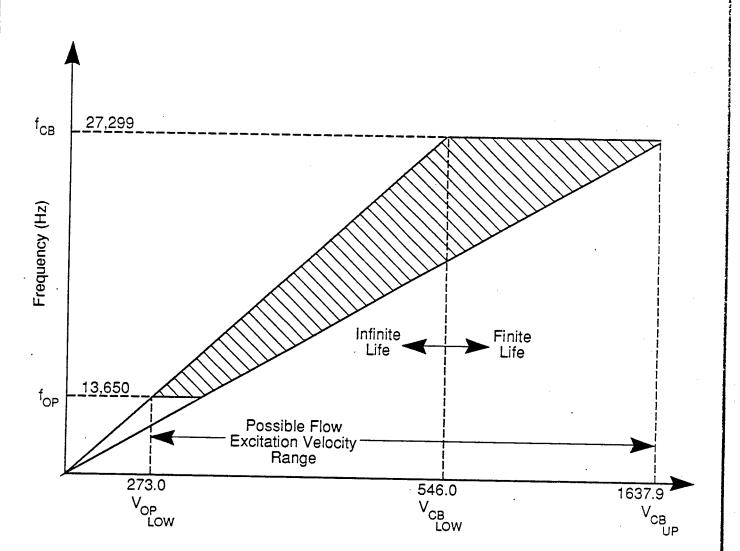
Table B-2. Flexhose Example Summary

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Fluid Velocity (FT/SEC)

Figure B-2. Frequency vs. Velocity Plot Indicating Flow Excitation Range for Example 2.0

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APPENDIX C

BELLOWS FLOW-INDUCED VIBRATION COMPUTER PROGRAM

"BELFIV" Version 3.3

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1.0 COMPUTER PROGRAM:

The computer program was written to calculate the modal frequencies, flow excitation ranges, and flow-induced stresses for the longitudinal modes and the convolute bending mode in a metal bellows. This program is written to apply to both liquid and gas flows through a metal bellows. In the case of gas flows, it also calculates the first radial acoustic mode frequency and velocity. This program, however, does not conduct flow-induced vibration analysis for a flexhose and does not calculate static stresses in a bellows. The flexhose and static stress analysis have to be done by hand.

This computer program takes into account the uncertainty factors and acoustic factor for flow-induced stress. The output values for flow-induced stress have the appropriate uncertainty factors and acoustic factor already applied.

The computer program gives the user the option of inputing the data from the keyboard or from a data file. An example of an input data file is given in paragraph 1.2. The output data can be saved in a data file or sent to a printer.

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The computer program was written in Fortran language to run interactively with an IBM-XT personal computer. However, this does not limit its use as it can be readily modified to suit the needs of the designer. The differences between the hand calculations and computer calculations are attributed to round-off errors. Differences may also be found when different Fortran compilers are used. The Fortran compiler used for this computer program is the IBM Professional Fortran Compiler, version 1.00 by Ryan-McFarland Corp.

1.1 <u>Comparison of Theoretical and Computer Program Variables:</u>

ANALYSIS	COMPUTER	COMMENT
a	A	Mean convolute radius
$\mathtt{B}_{\mathtt{N}}$	BN	Dimensionless frequency factor
c _E	CE	Elbow factor
C _{NP}	CNP	Damping modifier coefficient
c*	CST	Force and damping coefficient
Cø	CO	Speed of sound
D _i	DI	Bellows inside diameter
Do	DO	Bellows outside diameter
$D_{\mathbf{m}}$	DMEAN	Bellows mean diameter
E .	E	Young's modulus of elasticity

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				, , , , ,
fa	FREQCO	First radial ac	coustic	mode
fo	FO	Reference frequ	lency	
f _C	FREQC	Critical freque	ency for	.
f _{CB}	FREQCB	Convolute bendi frequency	.ng mode	2
f(N)	FREQ(MODE)	Modal frequency	•	
a	G	Gravitational a	ccelera	tion
h	Н	Mean inside con	volute	height
k	K	Elemental sprin	g rate	
Ka	KA	Overall bellows	spring	rate
m	MASS MASSR	Total elemental	mass	
$m_{ m m}$	MMETAL	Elemental metal	mass	
^m fl	FLUID1	Fluid added mas	s	
^m f2	FLUID2	Fluid added mas	s	
^m f	MFLUID MFLUDR	Total elemental added mass	fluid	
N	MODE	Mode number		
2N _C -1	NDEG	Number of degre	es of	
		freedom for a b	ellows	
N _C	NC	Number of convo	lutes	
Np	NPLY	Number of plys		
P_{D}	PD	Free stream dyn	amic pr	essure
P	P	Fluid pressure		
		•		
		•		

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s _{o1}	STLO	Lower Strouhal number
Sou	STUP	Upper Strouhal number
Soc	STCRIT	Critical Strouhal number
t	T	Ply thickness
$v_{ t low}$	V(MODE,1)	Lower limit velocity for mode N
v*	V(MODE,2)	
V _{up}	V(MODE,3)	Upper limit velocity for mode N
v _c	VELC	Critical velocity for mode N=Nc
v ′	VP	Normalized velocity parameter
[∨] 2_	VELCO	First radial acoustic mode velocity
X	GAMMA	Specific heat ratio for the gas
0	SIGMA	Inside convolute width
δ	DELTA	Inside convolute gap
2	LAMBDA	Inside convolute pitch
$\mathcal{P}_{\mathtt{f}}$	RHOF	Weight density of fluid
$\mathcal{P}_{\mathtt{m}}$	RHOM	Weight density of flexible line material.

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1.2 <u>Input Data File Format and Examples</u>

Input File: DFILE

Description of Input File DFILE:

Line 1--TITLE (A70)

This line assigns an identifying label to a particular bellows.

Line 2--JFLAG, NFLUID, NDEG (313)

JFLAG=1

This line enters certain conditions of the bellows.

JFLAG determines origin of overall spring rate KA

. program will calculate KA

NFLUID flow medium

NFLUID=2 liquid

NFLUID=1 gas

NDEG no. of bellows
longitudinal degrees of
freedom=2NC-1

DWG

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Line 3NC, NPLY,	SIGMA	., I	AMBDA, H	, T (6F1)	0.31	
This line ent						
bellows.		- 2	rrii deome	cric para	meters	of the
NC	• •	• •	. number from t	of convo	lutes d	counted
NPLY	• •	• •	. number	of plys		
				convolut		
LAMBDA .				convolut		
н	• • •	•	. mean in inches	nside con	volute	height,
T		•	. ply th	ickness,	inches	
Line 4DI, DO, E,	RHOM,	KA	, LOVERD	(2F10.3,	F10.0,	3F10.3)
This line ente	rs ce	rtai	n geomet	ric param	neters,	matl.
properties, an						
DI	• • •		bellows	inside d	iameter	, inches
DO	• • •			outside		4
E	• • •	• •	Young's lbs/sq.	modulus in.	of elas	ticity,
RHOM	• •	• •	weight o	density of lensity of	f the in.	
ка	• •	• •	bellows	overall s	enring :	rate, program
•						

CONTINUATION SHEET REVISIONS SYM DESCRIPTION DATE APPROVAL LOVERD length from termination of elbow to first convolute divided by the I.D. of pipe just before the bellows (input 0.0 if no elbow upstream) Line 5--This line enters the conditions of the fluid whether a gas or a liquid. For a liquid--P, TEMP, RHOF (3F10.3) liquid pressure, psig liquid temperature, Faherenheit Weight density of liquid at P and TEMP, lbf/cu. ft. gas pressure, psig gas temperature, Faherenheit gas pressure at reference

For a gas--P, TEMP, PREF, TREF, RHOREF (5F10.4)

TEMP

RHOF

TEMP

PREF state, psia

TREF gas temperature at reference state, Fahrenheit

RHOREF Weight density of gas at reference state, lbf/cu. ft.

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Line 6--Z, ZREF, GAMMA (3F10.3)

This line enters the gas compressibility factors and specific heat ratio for the gas. This line is only used when the fluid is a gas and is not used when the fluid is a liquid.

Z	•	•	•	•	•	•	•	•	•	•	•	<pre>gas compressibility (non-dim)</pre>	factor
---	---	---	---	---	---	---	---	---	---	---	---	--	--------

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An example of an input file for liquid flow through a metal bellows is shown below. The corresponding output file is shown in paragraph 1.3. This example is identical to the liquid medium example 1.1 presented in Appendix B.

LIQUID	MEDIUM	EXAMPLE	1.1			
1 2	31					
16.	.000	3.000	0.095	0.148	0.325	0,007
3.	.000	3.690	29000000.	0.286	0.000	1.333
35.	.000	68.000	62.400			

An example of an input file for gas flow through a metal bellows is shown below. The corresponding output file is shown in paragraph 1.3. This example is identical to the gaseous medium example 1.2 presented in Appendix B.

GASEOUS MEDIUM	EXAMPLE 1.2		•		
1 1 13					
7.000	1.000	0.400	0.726	1.250	0.037
8.000	10.574	28500000.	0.282	0.000	0.000
39.3000	-200.0000	14.7000	68.0000	0.0730	
0.982	1.000	1.400			

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1.3 Output File Examples

On the following pages are the output files corresponding to the two sample input files (liquid and gas).

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LIQUID MEDIUM EXAMPLE 1.1

BELLOWS PARAMETERS

SIGMA(INSIDE CONVOLUTE WIDTH, IN)	0.095
LAMBDA (INSIDE CONVOLUTE PITCH, IN)	0.148
H(MEAN INSIDE CONVOLUTE HEIGHT, IN)	0.325
T(CONVOLUTE THICKNESS PER PLY, IN)	0.007
DI(INSIDE DIAMETER, IN)	3.000
DO(OUTSIDE DIAMETER, IN)	3.690
NC(NUMBER OF CONVOLUTES)	16.000
NPLY(NUMBER OF PLYS)	3.000
	900E+08
	181.735
RHOM(MATERIAL DENSITY, LBF/CU.IN)	0.286

FLUID PARAMETERS

P(PRESSURE, PSIG)	35.000
TEMP(TEMPERATURE, DEG F)	68.000
RHOF(FLUID DENSITY, LBF/CU.IN)	0.3611E-01
NFLUID(1=GAS, 2=LIQUID)	2
CE(ELBOW FACTOR, DIMENSIONLESS)	2.410

THEORETICAL BELLOWS PERFORMANCE

LONG.	FLOW-IND. STRESS	MODE FREQUENCY	FLOW EXCI	TATION RANGE,	FT/SEC
MODE NO.	WITH U.F., PSI	HZ	LOWER	CRITICAL	UPPER
1	0.10502E+05	135.638	3.579	5.369	10.738
2	0.21952E+05	256.980	6.781	10.172	20.344
3	0.33304E+05	366.715	9.677	14.516	29.032
4	0.43439E+05	466.738	12.317	18.475	36.950
5	0.51738E+05	558.422	14.736	22.104	44.208
6	0.58014E+05	642.788	16.962	25.444	50.887
. 7	0.62338E+05	720.613	19.016	28.524	57.049
8	0.64914E+05	792.497	20.913	31.370	62.739
9	0.66006E+05	858.912	22.666	33.999	67.997
10	0.65896E+05	920.234	24.284	36.426	72.852
11	0.64862E+05	976.768	25.776	38.664	77.327
12	0.63155E+05	1028.764	27.148	40.722	81.444
13	0.61000E+05	1076.429	28.406	42.609	85.217
14	0.58585E+05	1119.936	29.554	44.331	88.662
15	0.56062E+05	1159.435	30.596	45.894	91.789
16	0.53554E+05	1195.053	31.536	47.304	94.608
17	0.56752E+05	1226.904	32.377	48.565	97.130
18	0.59664E+05	1255.088	33.120	49.681	99.361
19	0.62256E+05	1279.697	33.770	50.655	101.309
20	0.64508E+05	1300.814	34.327	51.491	102.981
21	0.66410E+05	1318.519	34.794	52.191	104.383
22	0.67960E+05	1332.886	35.173	52.760	105.520
23	0.69160E+05	1343.988	35.466	53.200	106.399
24	0.70015E+05	1351.896	35.675	53.513	107.025
25	0.70532E+05	1356.679	35.801	53.702	107.404
26	0.70719E+05	1358.407	35.847	53.770	107.541
27	0.70583E+05	1357.149	35.814	53.720	107.441
28	0.70132E+05	1352.978	35.704	53.555	107.111
29	0.69374E+05	1345.964	35.518	53.278	106.555
30	0.68316E+05	1336.180	35.260	52.890	105.781
31	0.66969E+05	1323.703	34.931	52.397	104.793
CONVOLUTE BENDING					

BENDING

MODE 0.30653E+06 2440.707

64.408 96.611

193.223

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GASEOUS MEDIUM EXAMPLE 1.2

BELLOWS PARAMETERS

FLUID PARAMETERS

THEORETICAL BELLOWS PERFORMANCE

LONG. MODE NO. 1	FLOW-IND. STRESS WITH U.F., PSI 0.52738E+03	MODE FREQUENCY HZ	FLOW EXC	ITATION RANG CRITICAL	GE, FT/SEC UPPER
2 3 4 5 6 7 8 9 10 11 12	0.10446E+04 0.14377E+04 0.16493E+04 0.16705E+04 0.15359E+04 0.13086E+04 0.16087E+04 0.18747E+04 0.20817E+04 0.16690E+05 0.17352E+05 0.17690E+05	122.691 243.368 360.526 472.710 578.533 676.693 765.990 845.336 913.776 970.493 1014.819 1046.246	13.632 27.041 40.058 52.523 64.281 75.188 85.110 93.926 101.531 107.833 112.758 116.250 118.270	20.449 40.561 60.088 78.785 96.422 112.782 127.665 140.889 152.296 161.749 169.137 174.374	40.897 81.123 120.175 157.570 192.844 225.564 255.330 281.779 304.592 323.498 338.273 348.749 354.809
CONVOLUTE BENDING MODE	0.52836E+05	1535.182	170.576	255.864	511.727

FIRST RADIAL ACOUSTIC MODE FREQUENCY= 980.654 HZ

FIRST RADIAL ACOUSTIC MODE VELOCITY= 163.442 FT/SEC

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Program Listing for BELFIV

THIS IS THE BELFIV PROGRAM ----- VERSION 3.3 THIS PROGRAM CALCULATES THE MODAL FREQUENCIES, FLOW EXCITATION C RANGES, AND THE FLOW-INDUCED STRESSES FOR THE LONGITUDINAL C MODES AND THE CONVOLUTE BENDING MODE IN A METAL BELLOWS. C THIS PROGRAM APPLIES TO BOTH LIQUID AND GAS FLOWS. C CASE OF GAS FLOWS, IT ALSO CALCULATES THE FIRST RADIAL ACOUSTIC C MODE FREQUENCY AND VELOCITY. THIS PROGRAM, HOWEVER, DOES NOT С CONDUCT FLOW-INDUCED VIBRATION ANALYSIS FOR A FLEXHOSE AND C DOES NOT CALCULATE STATIC STRESSES IN A BELLOWS. C C THIS PROGRAM WAS WRITTEN TO OPERATE ON AN IBM-XT COMPUTER C WITH A FORTRAN COMPILER WRITTEN BY RYAN-MCFARLAND CORPORATION. C THE FOLLOWING PARAMETERS ARE USED: C C JFLAG = 1(COMPUTE KA), 2(USE GIVEN KA). KA IS THE OVERALL C BELLOWS SPRING RATE, LBF/IN NFLUID = 1(GAS), 2(LIQUID)
NDEG = NUMBER OF BELLOWS LONGITUDINAL DEGREES OF FREEDOM, 2*NC-1 C C NC = NUMBER OF BELLOWS CONVOLUTES COUNTED FROM THE OUTSIDE SIGMA = INSIDE CONVOLUTE WIDTH, IN. С LAMBDA = INSIDE CONVOLUTE PITCH, IN. C H = MEAN INSIDE CONVOLUTE HEIGHT, IN. C T = CONVOLUTE THICKNESS PER PLY, IN. NPLY = NUMBER OF PLYS IN THE BELLOWS CONVOLUTES C DI = BELLOWS INSIDE DIAMETER, IN. C DO = BELLOWS OUTSIDE DIAMETER, IN. C E = YOUNG'S MODULUS OF THE BELLOWS MATERIAL, LB/SQ IN. C RHOM = WEIGHT DENSITY OF THE BELLOWS MATERIAL, LBF/CU IN. C LOVERD = LENGTH FROM TERMINATION OF ELBOW TO FIRST CONVOLUTE

DIVIDED BY THE I.D. OF PIPE JUST BEFORE THE BELLOW, NON-DIM. CE = DIMENSIONLESS ELBOW FACTOR

IF NFLUID = 1(GAS), THE PERFECT GAS EQUATION OF STATE IS USED FOR CALCULATING GAS DENSITY AT THE STATE DEFINED BY P AND TEMP. IT IS ASSUMED THAT THE GAS PROPERTIES ARE KNOWN AT A REFERENCE STATE DEFINED BY RHOREF, PREF, AND TREF.

P = GAS PRESSURE, PSIG TEMP = GAS TEMPERATURE, DEG. F.

PREF AND TREF = REFERENCE GAS STATE, PSIA AND DEG. F. RHOREF = WEIGHT DENSITY OF GAS AT REFERENCE STATE, LBF/CU FT.

Z = GAS COMPRESSIBILITY FACTOR, NON-DIM.

ZREF = GAS COMPRESSIBILITY FACTOR AT REFERENCE STATE, NON-DIM.

GAMMA = SPECIFIC HEAT RATIO FOR THE GAS, NON-DIM. IF NFLUID = 2(LIQUID), THE LIQUID DENSITY MUST BE KNOWN APRIORI AT THE LIQUID STATE (P AND TEMP). P = LIQUID PRESSURE, PSIG

TEMP = LIQUID TEMPERATURE, DEG. F.

RHOF = WEIGHT DENSITY OF LIQUID AT P AND TEMP, LBF/CU FT.

IMPLICIT REAL(A-H, O-Z) REAL MODER, MASS, MFLUID, MFLUDR, MMETAL, MASSR REAL KA, K, N1, LOVERD, NC, NPLY, LAMBDA, FLUID1, FLUID2 INTEGER*2 ANS, DEV CHARACTER*5 TITLE(80), DFILE(20), OFILE(20)

DIMENSION FREQ(75), V(75,3), FISC(75)

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	SYM	DESCRIPTION	DATE
C			
C SET DATA FILES			
c .			
WRITE(6,5)			
5 FORMAT(1X, 'DID YOU SET OUT: \$' (YES=1, NO=2)'/)	ONA TUP	TNDUT STIR	
\$' (YES=1, NO=2)'/)		INFOI FILE NAMES ?'/	
READ(5,*) NAM			
GO TO (20,10), NAM 10 WRITE(6,15)			
15 FORMAT(1X /PETTIDN TO DOG -			
15 FORMAT(1X, 'RETURN TO DOS EN S' (SET DFILE=input.DAT)'/' GO TO 2000	VIRONME	NT TO SET FILE NAMES. 1/	
GO TO 2000	(SET OF	'ILE=output.DAT)') '	
20 WRITE(6.25)			
25 FORMAT(1X, WILL THE INPUT D \$' KEYBOARD=1,FILE=2'/)	ATA BE	FROM TUE VEVECASE	
READ(5,*) INP		THE RETBOARD OR FI	LE ?'/
GO TO (30,220), INP			
C (30,220),1NP			
C INPUT DATA FROM KEYBOA	DD		
•	KD		
30 WRITE(6,35)			
35 FORMAT(1X, 'HOW DO YOU IDENT READ(5,40)(TITLE(1), J=1,70)	IFY THIS	S BELLOWS 24 /	
READ(5,40)(TITLE(I),I=1,70) 40 FORMAT(70A1)		22220NS : ,/)	
WRITE(6,45)			
45 FORMAT (1X. COMPUTE OF HOR			
45 FORMAT(1X, COMPUTE OR USE GI \$' 1 (COMPUTE KA), 2 (USE GIV READ(5.*) JF1AC	VEN SPR	RING RATE ?'/	
	EN KA)	/)	
GO TO (60,50), JFLAG			
50 WRITE(6,55)			
55 FORMAT(1X, 'OVERALL BELLOWS S READ(5,*) KA	PRING R	ATE, LBF/IN, 2//\	
60 CONTINUE		, , ,	
WRITE(6.65)			
65 FORMAT(1X, 'IS THE FILLED & CA	S OP TT	OUTD O	
	O OK LI	501D 3./	
READ(5,*) NFLUID			
WRITE(6,70)			
70 FORMAT(1X, 'NUMBER OF BELLOWS \$?'/)	CONVOLU	JTES COUNTED FROM THE OU	MCTDE
READ(5,*) NC		- 191911 11111 00	ISIDE
NDEG = 2*NC-1.			
WRITE(6,75)			
75 FORMAT(1X, NUMBER OF PLYS IN READ(5,*) NPLY	THE BEL	LOWS CONVOLUTES 344	
WRITE(6,80)		CONVOLUTES !!/)	*
80 FORMAT(1X, 'INSIDE CONVOLUTE W READ(5,*) SIGMA			
	IDTH, I	N. ?'/)	
WRITE(6.85)			
85 FORMAT(1X,'INSIDE CONVOLUTE P READ(5,*) LAMBDA	ITCH. T	N 2//	
READ(5,*) LAMBDA WRITE(6,90)	,	//	
90 FORMAT(1X /MEAN THETER			
90 FORMAT(1X, MEAN INSIDE CONVOLUTE READ(5, *) H	UTE HEI	GHT, IN. ?'/)	
WRITE(6.95)			
95 FORMAT (1X, CONVOLUTE THICKNESS	ים משם ?	'U T1	
READ(5,*) T	J LLK P	21, IN. ?'/)	
WRITE(6,100)			
100 FORMAT(1X, 'BELLOWS INSIDE DIAM READ(5,*) DI	ETER, I	IN. ?'/)	
WRITE(6.105)			
105 FORMAT(1X, BELLOWS OUTSIDE DIA	Venee		
COISIDE DIA	METER,	IN. ?'/)	
•			

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```
READ(5,*) DO
        WRITE(6,110)
    110 FORMAT(1X, 'YOUNGS MODULUS FOR BELLOWS MATERIAL, LB/SQ IN. ?'/)
        WRITE (6, 115)
    115 FORMAT(1X, 'WEIGHT DENSITY OF THE BELLOWS MATERIAL, LBF/CU IN. ?'/)
        READ(5,*) RHOM
        WRITE(6,120)
    120 FORMAT(1X, 'LENGTH FROM TERMINATION OF ELBOW TO FIRST CONVOLUTE'/
       S' DIVIDED BY THE I.D. OF PIPE JUST BEFORE THE BELLOW ? (INPUT O IF
        READ(5,*) LOVERD
        GO TO (125,160), NFLUID
    125 WRITE(6,130)
   130 FORMAT(1X, 'GAS PRESSURE, PSIG ?'/)
        READ(5,*) P
       WRITE (6, 135)
   135 FORMAT(1X, 'GAS TEMPERATURE, DEG. F ?'/)
       READ(5,*) TEMP
       WRITE(6,140)
   140 FORMAT(1X, 'GAS PRESSURE AT REFERENCE STATE, PSIA ?'/)
       READ(5,*) PREF
       WRITE(6,145)
   145 FORMAT(1X, 'GAS TEMPERATURE AT REFERENCE STATE, DEG. F ?'/)
       READ(5,*) TREF
       WRITE(6,150)
   150 FORMAT(1X, WEIGHT DENSITY OF GAS AT REFERENCE STATE, LBF/CU FT. ?
       READ(5,*) RHOREF
       WRITE(6,151)
   151 FORMAT(1X, 'GAS COMPRESSIBILITY FACTOR, NON-DIM. ?'/)
       READ(5,*) Z
       WRITE(6,152)
   152 FORMAT(1X, 'GAS COMPRESSIBILITY FACTOR AT REFERENCE STATE, NON-DIM.
      READ(5,*) ZREF
       WRITE(6,155)
  155 FORMAT(1X, 'SPECIFIC HEAT RATIO FOR THE GAS, NON-DIM. ?'/)
      READ(5,*) GAMMA
      GO TO 180
  160 WRITE(6,165)
  165 FORMAT(1X,'LIQUID PRESSURE, PSIG ?'/)
      READ(5, *) P
      WRITE(6,170)
  170 FORMAT(1X, 'LIQUID TEMPERATURE, DEG.F ?'/)
      READ(5,*) TEMP
      WRITE(6,175)
  175 FORMAT(1X, WEIGHT DENSITY OF LIQUID AT THE LIQUID STATE (P AND TEM
     $P), LBF/CU FT. ?'/)
      READ(5,*) RHOF
C..... SAVE INPUT DATA FROM KEYBOARD
  180 WRITE(6,185)
 185 FORMAT(1X,'DO YOU WISH TO SAVE INPUT DATA ? (YES=1, NO=2)'/)
      READ(5,*) NSAVE
      IF(NSAVE .EQ. 2) GO TO 250
     OPEN (UNIT=10, FILE='DFILE')
     WRITE(10,225) (TITLE(I), I=1,70)
     WRITE(10,230)JFLAG,NFLUID,NDEG
```

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```
WRITE(10,235)NC, NPLY, SIGMA, LAMBDA, H, T
        IF (JFLAG .EQ. 1) KA=0.0
        WRITE(10,236)DI,DO,E,RHOM,KA,LOVERD
        GO TO (200,210), NFLUID
    200 WRITE(10,241)P, TEMP, PREF, TREF, RHOREF
        WRITE(10,242)Z,ZREF,GAMMA
        GO TO 250
    210 WRITE(10,242)P, TEMP, RHOF
        GO TO 250
  C..... INPUT
                DATA FROM FILE
    220 OPEN (UNIT=7, FILE='DFILE')
        READ(7,225) (TITLE(I), I=1,70)
   225 FORMAT (70A1)
       READ(7,230)JFLAG,NFLUID,NDEG
   230 FORMAT(313)
       READ(7,235)NC, NPLY, SIGMA, LAMBDA, H, T
       READ(7,236)DI,DO,E,RHOM,KA,LOVERD
   235 FORMAT(6F10.3)
   236 FORMAT(2F10.3,F10.0,3F10.3)
       GO TO (240,245), NFLUID
   240 READ(7,241)P, TEMP, PREF, TREF, RHOREF
   241 FORMAT(5F10.4)
       READ(7,242)Z,ZREF,GAMMA
   242 FORMAT(3F10.3)
       GO TO 250
   245 READ(7,242)P, TEMP, RHOF
   250 CONTINUE
       PI=3.1415927
       G=32.174049
       DMEAN=(DI+DO)/2.0
       GO TO (400,405), JFLAG
C
C..... CALCULATION OF SPRING RATE
  400 KA=DMEAN*E*(NPLY/NC)*(T/H)**3
  405 K=2.*NC*KA
C
C..... CALCULATION OF METAL MASS AND FLUID MASS
C
      A=(SIGMA-T*NPLY)/2.
      MMETAL=PI*RHOM*T*NPLY*DMEAN*(PI*A+H-2.*A)/G
      GO TO (410,415), NFLUID
  410 RHOF=(RHOREF/1728.)*((P+14.7)/PREF)*((TREF+460.)/(TEMP+460.))*
     $(ZREF/Z)
      GO TO 420
  415 RHOF=RHOF/1728.
  420 FLUID1=PI*RHOF*DMEAN*H*(2.*A-T*NPLY)/(2.*G)
      DELTA=LAMBDA-SIGMA
      FLUID2=RHOF*DMEAN*(H**3)/(G*DELTA)
C
C..... CALCULATION OF CRITICAL FREQUENCY AND VELOCITY (AT MODE N=NC)
      STUP=0.3
      STLO=0.1
      STCRIT=0.2
     MODER=NC
     MFLUDR=1.0*FLUID1 + 0.68*(FLUID2*MODER)/NC
     MASSR=MFLUDR+MMETAL
```

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```
FO=(1./(2.*PI))*SQRT(12.*K/MASSR)
        FREQC=FO*SQRT(2.)
        VELC=FREQC*SIGMA/(STCRIT*12.)
  C
  C..... CALCULATION OF FREQ. AND VEL. RANGE FOR LOGITUDINAL MODES
 С
        DO 440 MODE=1, NDEG
       MFLUID=1.0*FLUID1 + 0.68*FLUID2*(MODE/NC)
       MASS=MFLUID+MMETAL
       BN=SQRT(2.*(1.+COS((PI*(2.*NC-MODE))/(2.*NC))))
       FO=(1./(2.*PI))*SQRT(12.*K/MASS)
       FREQ (MODE) = FO*BN
       DO 440 J=1,3
       GO TO (425,430,435),J
   425 V(MODE, J)=FREQ(MODE) *SIGMA/(STUP*12.)
       GO TO 440
   430 V(MODE, J) = FREQ(MODE) *SIGMA/(STCRIT*12.)
       GO TO 440
   435 V(MODE, J) = FREQ (MODE) *SIGMA/(STLO*12.)
   440 CONTINUE
 C
C..... CALCULATION OF FIRST RADIAL ACOUSTIC MODE (GAS MEDIA ONLY)
      GO TO (600,615), NFLUID
  600 RI=DI/2.
      HRI=H/RI
      CO=SQRT(GAMMA*(P+14.7)*G/(RHOF*12.))
      IF(HRI.LE.0.40) GO TO 605
      FNCO=-.336+.935*(RI/H)
      GO TO 610
  605 FNCO=3.8-16.72*(HRI**2)+13.67*(HRI**3)
  610 FREQCO=12.*FNCO*CO/(2.*PI*RI)
      QADJUS=5.0
      VELCO=FREQCO*SIGMA/(STCRIT*12.)
C
C..... CALCULATION OF FLOW-INDUCED STRESS FOR LONGITUDINAL MODES
  615 SSR=KA*NC/(DMEAN*NPLY)
      C1=.13
      C2 = .462
     C3=1.0
     C4=10.0
     C5=.06
     C6=1.25
     C7=5.5
     N1=1.0
     IF(LOVERD.EQ.0.0) N1=0.0
     CE=1.+(N1*4.7/(2.+LOVERD))
     DO 655 MODE=1,NDEG
     VP=V(MODE, 2)/VELC
     IF(NPLY.GT.1.) GO TO 620
     CNP=1.0
     GO TO 625
620 CNP=1.0-((C6*SIGMA/H)/(1.0+C7*VP**2))
625 BB=C1/(C2+VP**2)
     CC=C3*ABS(SIN(PI*VP))/(C4+VP**2)
     CST=BB+CC+C5
    PD=12.0*RHOF*(V(MODE,2)**2)/(2.0*G)
    DD=CST*T*PD/(VP*SSR*DELTA)
    EE=1.0+0.1*((400.0/SSR)**2)
```

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```
FIS=EE*DD*E*CNP*CE/NPLY
 C..... UNCERTAINTY FACTORS FOR STRESS (LONGITUDINAL MODES)
       GO TO (630,635), NFLUID
   630 IF (FREQ(MODE).GE.FREQCO) FIS=FIS*QADJUS*1.5
   635 CONTINUE
       GO TO (640,645), JFLAG
   640 FIS=FIS*2.0
       GO TO 650
   645 FIS=FIS*1.5
   650 CONTINUE
       FISC(MODE) = FIS
   655 CONTINUE
 C
C..... CALCULATION OF FREQ. AND VEL. RANGE FOR CONVOLUTE BENDING MODE
C
      FREQCB=(1./(2.*PI))*SQRT(8.*K*12./(MMETAL+.68*FLUID2))
       VCBLOW=FREQCB*SIGMA/(STUP*12.)
       VCBSTAR=FREQCB*SIGMA/(STCRIT*12.)
       VCBUP=FREQCB*SIGMA/(STLO*12.)
C..... CALCULATION OF FLOW-INDUCED STRESS FOR CONVOLUTE BENDING MODE
      VP=VCBSTAR/VELC
      IF (NPLY.GT.1) GO TO 660
      CNP=1.0
      GO TO 665
  660 CNP=1.0-((C6*SIGMA/H)/(1.0+C7*VP**2))
  665 CST=0.4
      PD=12.0*RHOF*(VCBSTAR**2)/(2.0*G)
      DD= CST*T*PD/(VP*SSR*DELTA)
      FIS=EE*DD*E*CNP*CE/NPLY
C..... UNCERTAINTY FACTORS FOR STRESS (CONVOLUTE BENDING MODE)
      GO TO (670,675), NFLUID
  670 IF (FREQCB.GE.FREQCO) FIS=FIS*QADJUS*1.5
  675 CONTINUE
      GO TO (680,685), JFLAG
  680 FIS=FIS*2.0
      GO TO 690
  685 FIS=FIS*1.5
  690 CONTINUE
      FISCB=FIS
C
C..... OUTPUT DATA
      DEV=6
 800 WRITE(DEV,805)(TITLE(I), I=1,70)
 805 FORMAT(1X,70A1)
     WRITE(DEV,840) SIGMA, LAMBDA, H, T, DI, DO, NC, NPLY, E
     WRITE(DEV, 845) KA, RHOM, P, TEMP, RHOF, NFLUID, CE
     WRITE (DEV, 850)
     DO 810 MODE=1, NDEG
 810 WRITE(DEV,855) MODE, FISC(MODE), FREQ(MODE), V(MODE, 1), V(MODE, 2),
    $V(MODE,3)
     WRITE (DEV, 856) FISCB, FREQCB, VCBLOW, VCBSTAR, VCBUP
     GO TO (815,820), NFLUID
 815 WRITE(DEV,860) FREQCO
```

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WRITE(DEV, 865) VELCO
  820 CONTINUE
  825 IF (DEV.EQ.8 .OR. DEV.EQ.9) GO TO 835
      WRITE(6,830)
  830 FORMAT(//,1X,'WHERE DO YOU WANT OUTPUT SENT? 0-EXIT,1-FILE,2-PRINT
     $ER'/)
      READ(5, *) ANS
      IF(ANS.EQ.0) GO TO 2000
      IF(ANS.EQ.1) DEV=8
      OPEN (UNIT=8, FILE='OFILE')
      IF(ANS.EQ.2) DEV=9
      OPEN (UNIT=9, FILE='LPT1')
      GO TO 800
  835 CONTINUE
  840 FORMAT(/,29X,18HBELLOWS PARAMETERS,//
             19X,33HSIGMA(INSIDE CONVOLUTE WIDTH, IN),4X,F6.3,/
             19X,34HLAMBDA(INSIDE CONVOLUTE PITCH, IN),3X,F6.3,/
             19X,35HH (MEAN INSIDE CONVOLUTE HEIGHT, IN),2X,F6.3,/
             19X,34HT(CONVOLUTE THICKNESS PER PLY, IN),3X,F6.3,/
             19X,23HDI(INSIDE DIAMETER, IN),14X,F6.3,/
             19X,24HDO(OUTSIDE DIAMETER, IN),13X,F6.3,/
             19X,24HNC(NUMBER OF CONVOLUTES),12X,F7.3,/
             19X,20HNPLY(NUMBER OF PLYS),16X,F7.3,/
             19X,28HE(YOUNG'S MODULUS, LB/SQ.IN),4X,E11.4)
 845 FORMAT(19X,31HKA(OVERALL SPRING RATE, LBF/IN),1X,F11.3,/
             19X,33HRHOM(MATERIAL DENSITY, LBF/CU.IN),3X,F7.3,//
             30X,16HFLUID PARAMETERS,//
             19X,17HP(PRESSURE, PSIG),19X,F7.3,
            19X,24HTEMP(TEMPERATURE, DEG F),11X,F8.3,/
             19X,30HRHOF(FLUID DENSITY, LBF/CU.IN),2X,E11.4,/
            19X,23HNFLUID(1=GAS, 2=LIQUID),19X,I1,/
            19X, 'CE(ELBOW FACTOR, DIMENSIONLESS)', 6X, F6.3///
            25X, 'THEORETICAL BELLOWS PERFORMANCE',/)
 850 FORMAT(2X,78HLONG.
                            FLOW-IND. STRESS
                                                 MODE FREQUENCY
    $ EXCITATION RANGE, FT/SEC,/,1X,8HMODE NO.,3X,14HWITH U.F., PSI,
                                                                      FLOW
    $11X,2HHZ,13X,5HLOWER,5X,8HCRITICAL,4X,5HUPPER,/)
 855 FORMAT(3X,I2,8X,E11.5,7X,F11.3,5X,3F11.3)
 856 FORMAT(//,1X,9HCONVOLUTE,/,2X,7HBENDING,/,3X,4HMODE,6X,E11.5,7X, $F11.3,5X,3F11.3)
 860 FORMAT(//,3X,'FIRST RADIAL ACOUSTIC MODE FREQUENCY=',F9.3,1X,
    $'HZ'/)
865 FORMAT(3X, 'FIRST RADIAL ACOUSTIC MODE VELOCITY=', F9.3, 1X,
    $'FT/SEC'/)
2000 CONTINUE
     END
```

DWG

From:

Ballentine, Laura A. (MSFC-ISO2)[MITS II]

Sent:

Thursday, October 31, 2019 7:48 AM

Moore, Marjorie I. (MSFC-ISO2)[MITS II]

Subject: RE: 20M02540, Rev. E

Attachments: FOIA 20-MSFC-F-00037 Search Form.pdf

Probably has something to do with the encryption.

Attached is the FOIA Search form with Sean's signature stating the document isn't restricted.

From: Moore, Marjorie I. (MSFC-ISO2)[MITS II] < marjorie.moore@nasa.gov>

Sent: Thursday, October 31, 2019 7:43 AM

To: Ballentine, Laura A. (MSFC-ISO2)[MITS II] < <u>laura.a.ballentine@nasa.gov</u>>

Cc: Grady, Alan (MSFC-ISO2)[MITS II] <alan.e.grady@nasa.gov>

Subject: RE: 20M02540, Rev. E

I was not able to open the attachment you send.



Marja Moore Repository Supervisor Bldg. 4491/Room 101 Phone (256) 544-4278



From: Ballentine, Laura A. (MSFC-ISO2)[MITS II] < <u>laura.a.ballentine@nasa.gov</u>>

Sent: Wednesday, October 30, 2019 1:18 PM

To: Moore, Marjorie I. (MSFC-ISO2)[MITS II] < <u>marjorie.moore@nasa.gov</u>>

Cc: Grady, Alan (MSFC-ISO2)[MITS II] <alan.e.grady@nasa.gov>

Subject: 20M02540, Rev. E

Marja,

Just wanted to make you aware that MSFC drawing 20M02540, Rev E was recently requested through FOIA. The electronic copy of this document is in the Repository's projects folder and the restriction indicates ITAR. As part of the response process, Sean Benson/AS50 in export control reviewed the drawing and determined that it is not subject to any restrictions including ITAR. (see attached email)

A suggested solution would be to remove the attached 2896 and change the drawings restriction property to 'None.' You can

determined to not be restricted.					
Thanks,					
Laura					

also include a statement in notes field indicating the drawing was reviewed by Sean Benson/AS50 on October 30, 2019 and

FOIA SEARCH INSTRUCTIONS

We identified your office as likely to maintain record(s) responsive to the below-referenced and attached FOIA request. Please review this request and complete a search, following the instructions detailed below. Please respond to this search request by the below-referenced due date. The due date allows the FOIA and Legal Offices adequate time to process and review records, and to complete a response to the FOIA request within the time period established by federal law.

Please conduct a search that is *reasonably calculated to locate all responsive records* described in the attached request. All responsive records <u>MUST</u> be provided to the <u>HQ FOIA OFFICE</u>. HQ FOIA must review each responsive record to determine whether it should be released or withheld either in part or in full. The records, even those withheld in full, are also required to be stored in the FOIA administrative file. If you have questions regarding production of a record, please notify our office immediately.

INSTRUCTIONS:

- 1. Subject Matter Expert:
 - · Conduct a search for existing records.
 - Please track how much time it takes (hours/minutes) to locate the records, and note that in the appropriate space below [NOTE: if it will take longer than 2 hours to complete the search, please notify the HQ FOIA Office prior to completing the search].
 - o Do not create or alter any records to satisfy a request.
 - NASA is not obligated to respond to questions, or add explanatory material to any records disclosed in response to a FOIA request.
 - o If no responsive records are located, please check the appropriate box below.
 - Provide the method and location of search. This information is required for the administrative file. For
 example, provide the names of individuals who searched; search terms used; indices, databases, and/or
 other record holdings that were searched; as well as any search methodologies used.
 - Download, copy, or scan the records.
 - Please do not provide original copies of records to the HQ FOIA Office.
 - o To the extent possible, it is helpful if electronic copies are provided.
 - Review the responsive records page by page and line by line, and <u>mark</u> any information that is not suitable for public release. You should also include a statement as to why the information should not be released.
 - You may highlight the record(s) <u>OR</u> place [red brackets] around the information, and note the FOIA exemption that applies, or your justification for protecting the information.
 - Do not black out any information. The FOIA office will do this once all records are reviewed and approved for release.
 - As you review the records, please confirm that all pages are included. If any part of the records are
 missing or if the number sequence changes, please provide a brief statement as to why the information is
 not included. Providing this information ahead of time will help to save processing time.
 - The legal requirement under FOIA is to provide all responsive records to our office even if they are not releasable.
 - If responsive records are located, you may recommend:
 - Full release Provide one clean copy only.
 - Partial release Provide a marked copy as described above AND a clean copy.
 - Withhold in full Provide a copy of the record clearly marked to withhold in full <u>AND</u> a clean copy.
 - Provide a brief statement for all withholding (partial/full) recommendations or "no records" notices. This statement will be used to conduct our analysis, prepare the response letter, and will also be part of the package sent to the legal office for their review and concurrence.
 - Return the records to our office with your withholding recommendations/justifications and completed FOIA Action Report (attached).
- 2. Export Control Representative (ECR), if applicable

If records located contain technical information, an export compliance review **MUST** be conducted before returning the documents to the FOIA Office. You should inform your CEA that you are reviewing a document that is expected to be released or withheld entirety under the FOIA. If any withholdings are taken for this type of information, the ECR must provide the pertinent statute that exempts the records from disclosure.

FREEDOM OF INFORMATION ACT (5 USC § 552) EXPLANATION OF EXEMPTIONS

- **(b)(1) Classified Material:** Applies to information that is currently and properly classified pursuant to Executive Order 13526 in the interest of national defense or foreign policy.
- **(b)(2)** Human Resource Internal Rules and Practices: consistent with the plain meaning of the term 'personnel rules and practices,' encompasses only records relating to issues of employee relations and human resources.

Exemption 2's New Three-part Test

- 1. The Information Must be Related to "Personnel" Rules and Practices As the Supreme Court emphasized, the "key word" in the exemption and the one word which "most clearly marks the provision's boundaries is 'personnel. The Department of Justice Guidance on Milner advises agencies that in order for information to qualify for protection under Exemption 2, agencies must ensure that the information at issue satisfies the requirement that it "relate to an agency's personnel rules or practices.
- 2. The Information <u>Must</u> Relate "Solely" to Those Personnel Rules and Practices The second requirement for Exemption 2 is that "the information at issue must <u>'relate solely' to the agency's personnel rules and practices.</u>" The Supreme Court defined this phrase by its "usual" meaning, which is "exclusively or only."
- 3. The Information <u>Must</u> be "Internal" The third requirement for Exemption 2 is that the <u>information must be</u> "'internal,' meaning that 'the agency must typically keep the records to itself for its own use. Exemption 2 is now limited to records that are <u>1) personnel-related rules and practices</u>; 2) that are "related solely" to such rules and practices; and 3) that are "internal" to the agency.83
- **(b)(3) Exempt by Specific Statute:** Applies to information specifically exempted by a statute establishing particular criteria for withholding. The language of the statute must clearly state that the information will not be disclosed. **(If this exemption is used the pertinent statute must be cited)**
- **(b)(4) Privileged/Confidential Commercial/Financial Information:** Applies to information such as trade secrets and commercial or financial information obtained from a company on a privileged or confidential basis, which if released would result in competitive harm to the company.
- **(b)(5)** Inter/Intra Agency Memoranda Deliberative in Nature: Applies to inter- and intra-agency documents which are both deliberative in nature and part of a decision making process; includes subjective evaluations, opinions and recommendations.
- **(b)(6) Privacy Act Sensitive Information:** Applies to information release of which would constitute a clearly unwarranted invasion of the personal privacy of individuals.
- **(b)(7) Information Compiled for Law Enforcement Purposes:** Applies to records or information compiled for law enforcement purposes that (a) could reasonably be expected to interfere with law enforcement proceedings; (b) would deprive a person of a right to a fair trial or impartial adjudication; (c) could reasonably be expected to constitute an unwarranted invasion of the personal privacy of others; (d) disclose the identity of a confidential source; (e) disclose investigative techniques and procedures, or (f) could reasonably be expected to endanger the life or physical safety of any individual.
- **(b)(8) Financial Institution Records:** Applies to records related to regulation or supervision of financial institutions.
- **(b)(9) Geological and Geophysical Information:** Applies to records relating to geological and geophysical information and data concerning to wells.
- <u>PLEASE NOTE</u>: Any reasonably segregable portion of a record shall be provided to any person requesting records after redacting portions that are exempt from release. The amount of information deleted, and the exemption under which the deletion is made, shall be indicated <u>on the released portion of the record</u>, unless including that indication would harm an interest protected by the exemption in this subsection under which the deletion is made. If technically feasible, the amount of the information deleted, and the exemption under which the deletion is made, shall be indicated <u>at the place in the record where such deletion is made</u>.

FOIA ACTION OFFICE REPORT

FOIA Case Number		Date of Searc	h Request		
		Search Due Date Date Returned to HQ FOIA			
HQ POC					
Searcher(s) and Search Ti	me:				
Name	GS Grade	Search Time (Hours/Minutes)	# Records Located	# Total Pages	Review Time (Hours/Minutes)
Search Terms Used:					
Systems/Databases sea	arched:				
 Search Results and Lead Responsive records located REVIEW TIME: Withholdings/redactions responsive records 	, reviewed for withl	-		o the above	instructions. TOTAL
□ Release in full recommen□ No responsive records locat					
☐ Please check this box if you		ogram office may m	aintain respons	sive records,	, and identify that
office. PROGRAM OFFICE:			·		
Signatui	e		Code		
Name	.		Date		
Action Office FOIA Coordi	nator Certificati	on:			
I reviewed this packet for complete based on information provided to m			A Office. To the	e best of my	knowledge, and
Signatu	re		Code		
Name			Date		

Additional Comments/Concerns:					