

A048



National Aeronautics and
Space Administration

MSFC-SPEC-626
FEBRUARY 28, 1990

George C. Marshall Space Flight Center
Marshall Space Flight Center, Alabama 35812

TEST CONTROL DOCUMENT FOR
ASSESSMENT OF FLEXIBLE LINES
FOR FLOW INDUCED VIBRATION

PREPARED BY:

PROPULSION LABORATORY
GEORGE C. MARSHALL SPACE FLIGHT CENTER

TEST CONTROL DOCUMENT FOR
ASSESSMENT OF FLEXIBLE LINES
FOR FLOW INDUCED VIBRATION



P. Tygielski

EP64

3-5-90

Prepared by:
Philip J. Tygielski
Control Mechanisms
and Propellant
Delivery Branch

Organization
EP64

Date

C. Cornelius

JCH

EP61

3-5-90

Approved by:
Charles S. Cornelius,
Chief, Component
Development Division

Organization
EP61

Date

J. McCarty

EP01

3/6/90

Approved by:
John P. McCarty,
Director, Propulsion
Laboratory

Organization
EP01

Date

Stress Approval

Materials Approval

PKA/ED25/3/19/90
Pravin K Aggarwal

Redone per PMM Memo for EIT21 3/20/90

Dynamics Approval

Harry J. Bandgren ED 23 / 3/20/90

REVISION LOG

REVISION LETTER	DATE OF REVISION	REVISED PAGES	DESCRIPTION
—	2/28/90		BASIC RELEASE

NOTE: The portion of this specification affected by the changes is indicated by a vertical line in the outer margins of the page.

TABLE OF CONTENTS

	<u>Page</u>
1.0 GENERAL	4
1.1 SCOPE	4
1.2 PURPOSE	4
2.0 TEST SPECIMENS	4
2.1 SELECTION	4
3.0 TEST REQUIREMENTS	4
3.1 GENERAL	4
3.2 NON-VACUUM JACKETED BELLOWS	4
3.3 FLOW MEDIA	4
3.4 FACILITIES	4
3.5 OPERATIONAL LIFE	5
4.0 TEST SET-UP	5
4.1 GENERAL	5
4.2 INSTRUMENTATION	5
4.2.1 STRAIN GAGE ATTACHMENT	5
5.0 TEST PROCEDURES	5
5.1 INSTALLATION	5
5.2 RESONANCE SEARCH	6
5.3 RESONANCE DWELL	6
5.4 FAILURE MONITORING	6
6.0 REPORTS	6
6.1 CONTENTS	6

LIST OF FIGURES

	<u>Page</u>
FIGURE 1 - STRAIN GAGE LOCATIONS FOR A FREE BELLOWS . . .	7
FIGURE 2 - STRAIN GAGE INSTALLATION ON A FLEX HOSE . . .	8

1.0 GENERAL

1.1 SCOPE. This specification outlines the methods and criteria to employ in flow testing bellows and flex hose assemblies.

1.2 PURPOSE. To establish a standard flow test procedure for detecting flow induced vibrations in a given bellows and flex hose assembly. The MSFC Drawing #20M02540 provides the analytical assessment for predicting cyclic stress, excitation frequencies, and resonant flow ranges.

2.0 TEST SPECIMENS

2.1 SELECTION. Specimens shall be randomly selected from a production run. Bellows and flex hose assemblies selected as representative of several assemblies for qualification purposes should represent equal or worst case conditions.

3.0 TEST REQUIREMENTS

3.1 GENERAL. All testing must be performed simulating actual operating life conditions (i.e., medium, temperature, internal and external pressure, etc.,) unless otherwise approved by MSFC.

3.2 NON-VACUUM JACKETED BELLOWS. When testing non-vacuum jacketed bellows with both a cryogenic flow media and an external ambient environment, provisions must be made to avoid the dampening effect of liquid air, frost, or slush on the convolutes. A helium purge environment may be used if heat transfer problems will not occur; otherwise, a vacuum environment (10^{-5} to 10^{-6} torr) must be used.

3.3 FLOW MEDIA. The bellows and flex hose design fluid media shall be used for flow testing and conform to all requirements for its intended operating condition unless otherwise approved by MSFC. If a substitute media is to be used, an additional analysis must be performed to verify the bellows or flex hose design integrity.

3.4 FACILITIES. The flow facilities shall be capable of flowing the service media under the bellows or flex hose actual operating conditions for the duration of four (4) times the operational life and be capable of flowing +/- 10 percent of the expected flow range.

3.5 OPERATIONAL LIFE. Operational life shall be defined as the time accrued when the number of missions is multiplied by the mission system operation time plus any ground checkout operations.

4.0 TEST SET-UP

4.1 GENERAL. The flow test set-up shall include the actual operating configuration assemblies and supports. The bellows and flex hose elements of the line assembly shall be deflected for worst case static loading. Installation of flow affecting hardware; such as orifices, transitions, valves, or bends, within ten (10) diameters upstream of the bellows is required.

4.2 INSTRUMENTATION. Each bellows and flex hose assembly shall be instrumented to measure cyclic stress levels and excitation frequencies without impairing freedom of movement or the integrity of the bellows assembly. Flowrate through the bellows and flex hose assembly shall also be accurately measured. Instrumentation shall conform to the state-of-the-art hardware, where practical and feasible. Due to the variety of methods that can be used to gather strain gage and accelerometer data, it will be left to the discretion of the testing organization, with MSFC approval, as to the method employed. The method chosen must adhere to the local codes and abide by current approved practices.

4.2.1 STRAIN GAGE ATTACHMENT. If strain gages are used, they may be attached using an adhesive such as epoxy (Eastman 910 or equivalent). Surface preparation generally requires an agent and light sandblasting. Location of the gages shall be as follows:

Free Bellows: Locate gages on the second convolute crown from the upstream end and on the second convolute crown from the downstream end. Also, locate gages at half the length of the convoluted section. See figure 1 for strain gage locations.

Flex hose: A wire braid cut-out is required to install strain gage (figure 2). Locate near upstream end for liquid flow and near downstream end for gas flow.

5.0 TEST PROCEDURES

5.1 INSTALLATION. With strain gages installed, deflect the bellows assembly (i.e. axial deflection, lateral off-set, and angulation) to worst case static loading seen during actual operating conditions.

5.2 RESONANCE SEARCH. With the bellows or flex hose installed to simulate worst case loading possible during operating conditions, conduct a resonance search by slowly varying the flow rate from zero flow to maximum operating flow plus 10%. Caution shall be exercised to vary the flow at a rate which will not allow resonance to go undetected. Substantiate the search by a downsweep through the same range of flows. Record frequencies, strain levels, and corresponding flow rates. The data from this search is then studied to determine the flow rate at which to perform the resonant dwell testing. If a flow resonance is not detected, continue the flow scan until four (4) times the operational life requirements are met, or a total scan time of 3 hours has been obtained.

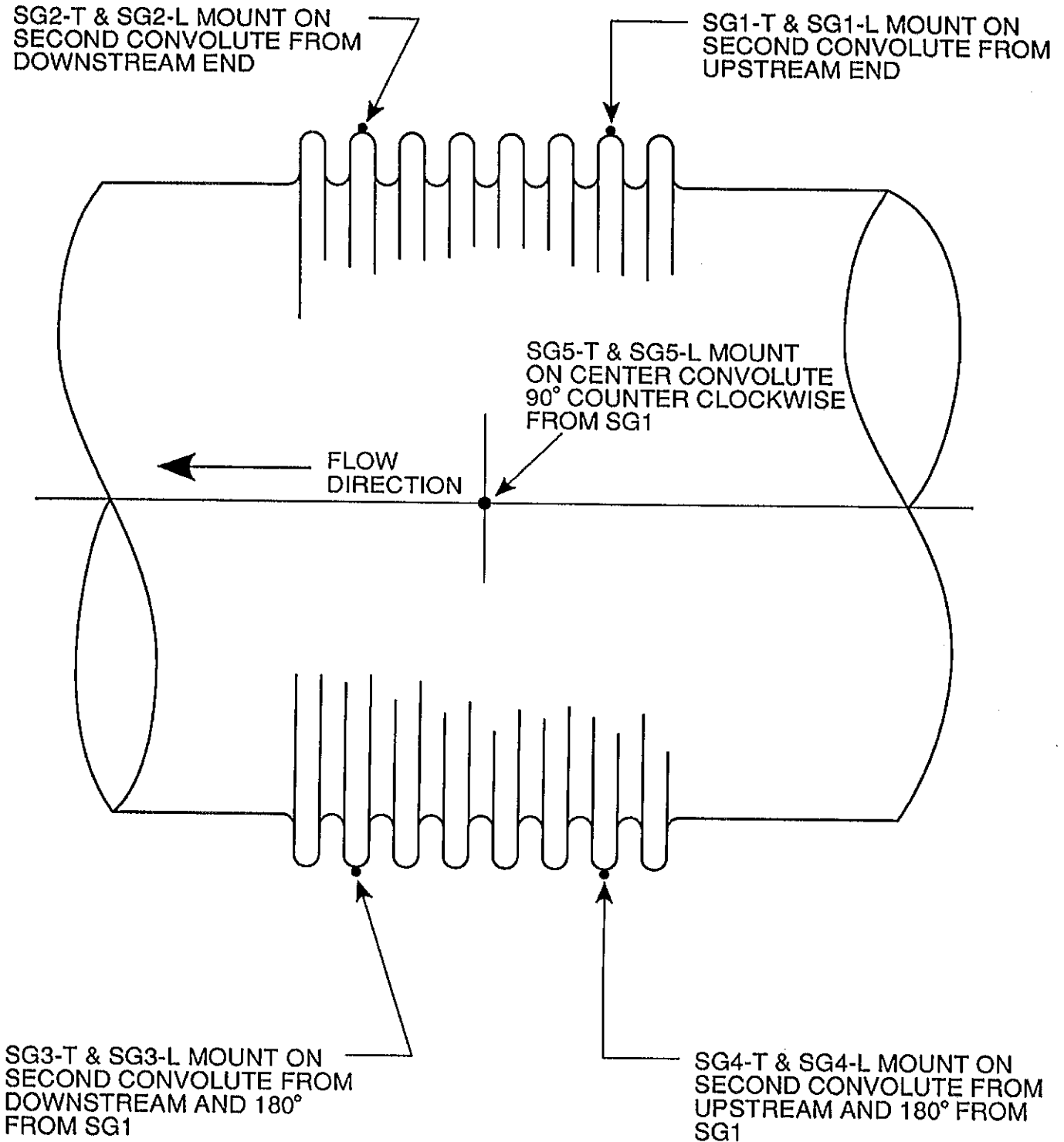
5.3 RESONANCE DWELL. This test will consist of dwelling at the most severe resonant conditions until the number of cycles equivalent to four (4) times the operational life have been accumulated at each resonance.

5.4 FAILURE MONITORING. A means to accurately determine the time to failure during the flow testing is required.

6.0 REPORTS

6.1 CONTENTS. Data shall include, but not be limited to, the following:

- a. Bellows/flex hose identification
- b. Measurement of as-built dimensions and a comparison to as-designed dimensions
- c. Measurement of actual spring rate
- d. Static conditions - pressure/deflection strains
- e. Dynamic conditions - resonance search data
 - test flow rates and pressures
 - time to failure or test duration
- f. Operational life requirements
- g. Sketch of test set-up or photograph
- h. Instrumentation utilized
- i. Temperature measurements



NOTES:

- T TANGENTIAL STRAIN GAGE
- L LONGITUDINAL STRAIN GAGE

FIGURE 1. STRAIN GAGE LOCATIONS FOR A FREE BELLOWS

TO RECORDING EQUIPMENT

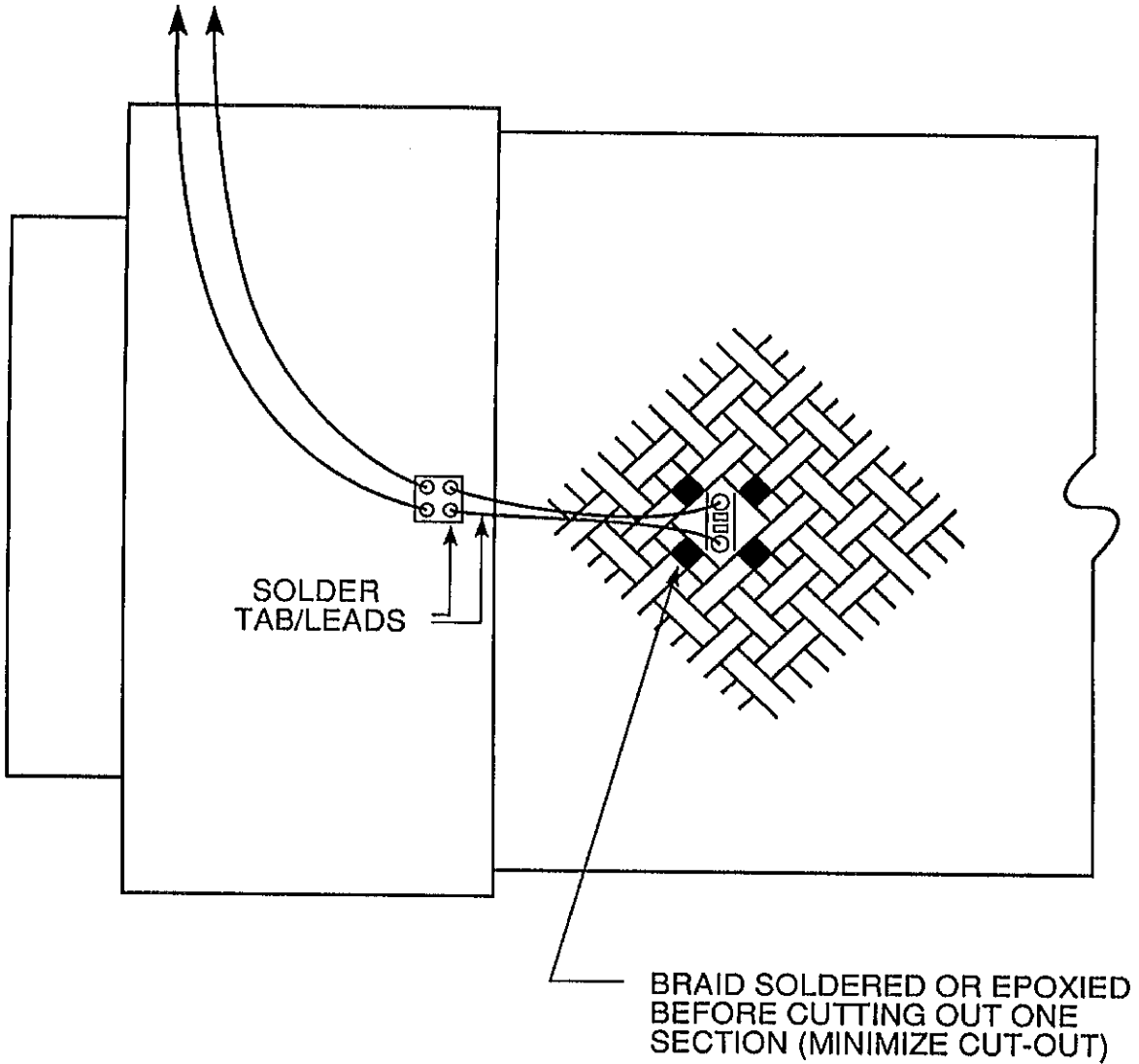


FIGURE 2. STRAIN GAGE INSTALLATION ON A FLEX HOSE

PACKAGE NO. 10443R

DOCUMENTATION RELEASE LIST
 GEORGE C. MARSHALL SPACE FLIGHT CENTER

MSFC CODE IDENT 14981/339B2
 ISSUE DATE FEB 22 2007

C H	DOCUMENT NUMBER	DRL DRL DSH REV	TITLE	CCBD NO.	PCN	PC	EFFECTIVITY
*	MSFC-SPEC-626	202 -	TEST CONTROL DOCUMENT FOR ASSESSMENT OF FLEXIBLE LINES FOR FLOW INDUCED VIBRATION	SB3-00-0000	0000000	A	NONE

CHG NO.	CHG REV	CHG NOTICE	RESPONSIBLE ENGINEER	RESPONSIBLE ORGANIZATION	ACTION DATE	DESCRIPTION
			P. TYGIELSKI	EP64	04/13/94	BASELINE RELEASE
*	1	SCN000	EUGENA GOGGANS	EO03	02/22/07	DOCUMENT RELEASED THRU PDS. NO LONGER TRACKED IN ICMS.

CHECKER

N/A
 02/15/07

(FINAL)

PACKAGE NO: 10443R

PROGRAM/PROJECT: MULTI

LAST UPDATED: 02/22/07

NOMENCLATURE: MSFC-STD- GOING TO NONE EFFECTIVITY

ECR NO:	PCN:	CCBD NO:	DATE PREPARED:
EO03-0000	0000000	000-00-0000 SB3-00-0000	02/22/07

DWG SIZE	DRAWING NUMBER	DWG REV	EPL/DRL/DDS NUMBER	DWG REV	EPL DSH	EPL REV	EO DASH NUMBER	EO REV	PART NUMBER
			MSFC-HDBK-1453		202	-			
			MSFC-HDBK-1674		202	-			
			MSFC-HDBK-2221		203	-			
			MSFC-HDBK-505		202	-			
			MSFC-HDBK-670		202	-			
			MSFC-MNL-1951		209	-			
			MSFC-PROC-1301		202	-			
			MSFC-PROC-1721		202	-			
			MSFC-PROC-1831		202	-			
			MSFC-PROC-1832		202	-			
			MSFC-PROC-404		202	-			
			MSFC-PROC-547		202	-			
			MSFC-QPL-1918		204	-			
			MSFC-RQMT-1282		202	-			
			MSFC-SPEC-1198		202	-			
			MSFC-SPEC-1238		202	-			
			MSFC-SPEC-1443		202	-			
			MSFC-SPEC-164		202	-			
			MSFC-SPEC-1870		202	-			
			MSFC-SPEC-1918		203	-			
			MSFC-SPEC-1919		206	-			
			MSFC-SPEC-2083		202	-			
			MSFC-SPEC-2223		202	-			
			MSFC-SPEC-2489		206	-			
			MSFC-SPEC-2490		205	-			
			MSFC-SPEC-2491		203	-			
			MSFC-SPEC-2492		203	-			
			MSFC-SPEC-2497		211	-			
			MSFC-SPEC-250		202	-			
			MSFC-SPEC-445		202	-			
			MSFC-SPEC-504		202	-			
			MSFC-SPEC-521		202	-			
			MSFC-SPEC-548		202	-			
			MSFC-SPEC-560		202	-			
			MSFC-SPEC-626		202	-			
			MSFC-SPEC-684		202	-			
			MSFC-SPEC-708		202	-			
			MSFC-SPEC-766		202	-			
			MSFC-STD-1249		202	-			
			MSFC-STD-1800		202	-			
			MSFC-STD-246		202	-			
			MSFC-STD-2594		203	-			

PACKAGE NO: 10443R

DWG SIZE	DRAWING NUMBER	DWG REV	EPL/DRL/DDS NUMBER	DWG REV	EPL DSH	EPL REV	EO DASH NUMBER	EO REV	PART NUMBER
			MSFC-STD-2903		202	-			
			MSFC-STD-2904		202	-			
			MSFC-STD-2905		202	-			
			MSFC-STD-2906		202	-			
			MSFC-STD-2907		202	-			
			MSFC-STD-366		202	-			
			MSFC-STD-383		202	-			
			MSFC-STD-486		202	-			
			MSFC-STD-506		203	-			
			MSFC-STD-531		202	-			
			MSFC-STD-557		202	-			
			MSFC-STD-561		203	-			
			MSFC-STD-781		202	-			

SUBMITTED BY ENGINEERING AREA:	BASIC	CHANGE	PARTIAL	COMPLETE	CLOSES	ACTION
EO03		X		X		EO03

PREPARED BY:
EUGENA GOGGANS
12/19/06

SUBMITTED BY:

CONCURRENCE:

TRANSMITTAL DATES
TO RELEASE DESK 02/22/07 10:00
TO MSFC DOC REP 02/22/07 00:00

REMARKS:

2007 FEB 22 AM 11:22

MSFC DOCUMENTATION REPOSITORY - DOCUMENT INPUT RECORD

I. GENERAL INFORMATION

1. APPROVED PROJECT: SPACE SHUTTLE	2. DOCUMENT/ DRAWING NUMBER: MSFC-SPEC-626	3. CONTROL NUMBER:	4. RELEASE DATE: 5/11/90	5. SUBMITTAL DATE:
6. DOCUMENT/DRAWING TITLE: TEST CONTROL DOCUMENT FOR ASSESSMENT OF FLEXIBLE LINES FOR FLOW INDUCED VIBRATION			7. REPORT TYPE: SPEC	
8. CONTRACT NUMBER / PERFORMING ACTIVITY: N/A		9. DRD NUMBER: N/A		10. DPD / DRL / IDRD NUMBER: N/A
11. DISPOSITION AUTHORITY (Check One): <input checked="" type="checkbox"/> Official Record - NRRS 8/12/1A <input type="checkbox"/> Reference Copy - NRRS 8/5/A/3 (destroy when no longer needed)		12. SUBMITTAL AUTHORITY: PHILIP TYGIELSKI		13. RELEASING AUTHORITY:
14. SPECIAL INSTRUCTIONS: N/A				
15. CONTRACTOR/SUBMITTING ORGANIZATION, ADDRESS AND PHONE NUMBER: N/A			16. ORIGINATING NASA CENTER: MSFC	
			17. OFFICE OF PRIMARY RESPONSIBILITY: TD62 SPACE TRANSPORTATION DIR.	
18. PROGRAMMATIC CODE (5 DIGITS): N/A			19. NUMBER OF PAGES: 10	

II. ENGINEERING DRAWINGS

20. REVISION: INITIAL	21. ENGINEERING ORDER:	22. PARTS LIST:	23. CCBD:
--	------------------------	-----------------	-----------

III. REPORTS, SPECIFICATIONS, ETC.

24. REVISION: INITIAL	25. CHANGE: —	26. VOLUME: —	27. BOOK: —	28. PART: —	29. SECTION: —
30. ISSUE: —	31. ANNEX:	32. SCN:	33. DCN:	34. AMENDMENT:	
35. APPENDIX: —	36. ADDENDUM: —	37. CCBD:	38. CODE ID: —	39. IRN:	

IV. EXPORT AND DISTRIBUTION RESTRICTIONS

<input type="checkbox"/> Privacy Act (see MWI 1382.1)	<input type="checkbox"/> EAR (see MPG 2220.1)
<input type="checkbox"/> Proprietary (see MPD 2210.1)	<input type="checkbox"/> Other ACI (see NPG 1620.1 and MPG 1600.1)
<input type="checkbox"/> Patent (see MPG 2220.1)	<input checked="" type="checkbox"/> No statutory or institutional restrictions applicable -- material may be electronically distributed to user in the NASA domain
<input type="checkbox"/> ITAR (see MPG 2220.1)	CER: H.J. Dennis, Jr. TD61

V. ORIGINATING ORGANIZATION APPROVAL

40. ORG. CODE: TD62	41. PHONE NUMBER: 544-7169	42. NAME: PHILIP TYGIELSKI	43. SIGNATURE/DATE: P. Tygielski 10/20/02
-------------------------------	--------------------------------------	--------------------------------------	---

VI. TO BE COMPLETED BY MSFC DOCUMENTATION REPOSITORY

44. RECEIVED BY: Jammy Wiso	45. DATE RECEIVED: 10-15-03	46. WORK ORDER:
---------------------------------------	---------------------------------------	-----------------